

Heat Sink Materials for the Plasma-facing Components of Fusion Devices

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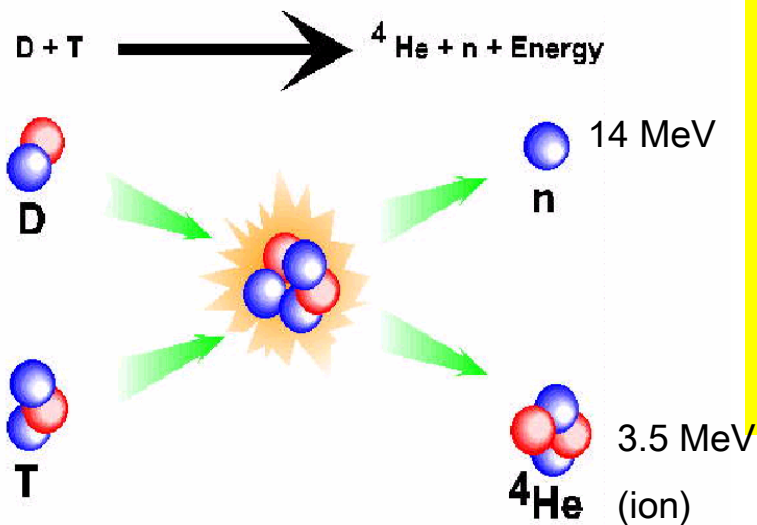
contents

- Fusion
- ITER: divertor requirements
 heat sink material: CuCrZr
- Reactor: divertor requirements →
- Cu-SiC MMC development
- Conclusions

Part of the work has been supported by the European Commission in the frame of the ExtreMat Integrated Project



D-T Fusion

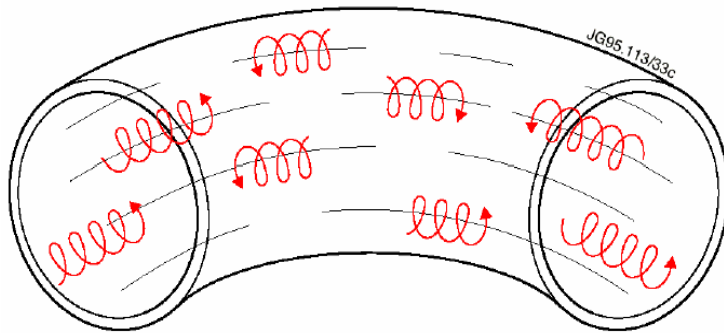


Parameter field for a fusion reactor

plasma density (n): $>10^{20} \text{ m}^{-3}$
plasma temperature (T): 18 keV
(equ. 200 Mio. deg.)
energy confinement time (τ): $>1 \text{ s}$

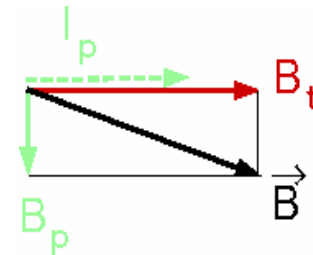
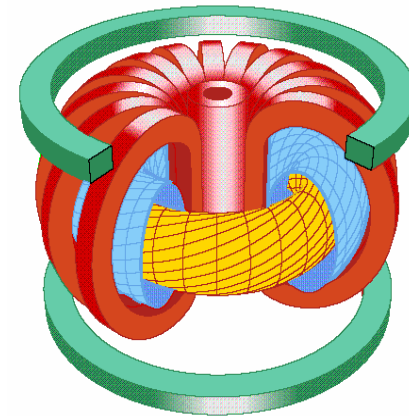
plasma volume: approx. 1000 m^3
fusion power: 2 GW

principle of toroidal magnetic confinement



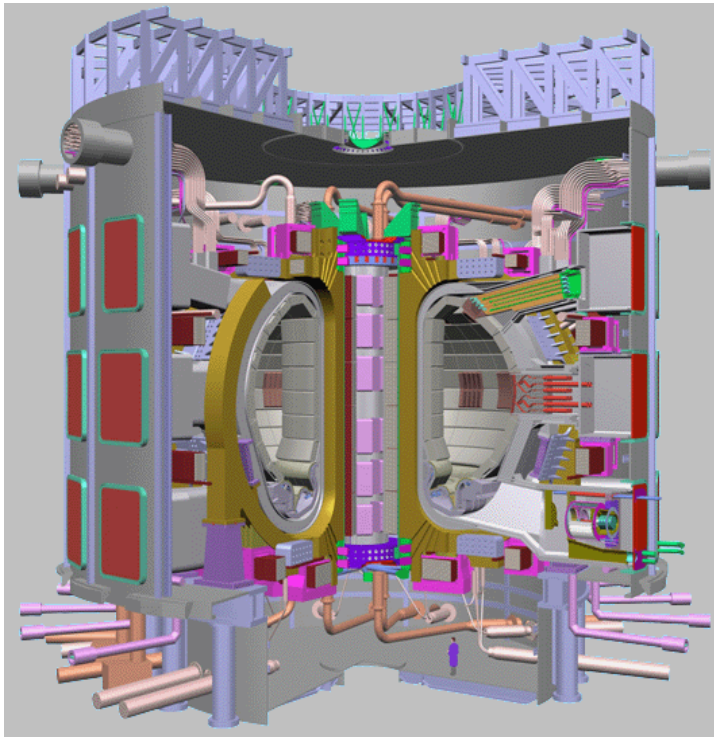
- magnetic field confines ions, electrons
- balances the plasma pressure (10 atm)
- thermal insulation (200 Million K)

Tokamak



Next step: ITER

ITER Design Parameters



Fusion Power	500 MW
Plasma Volume	837 m³
Plasma Surface	678 m²
Heat flux on Divertor	10 (20) MW/m²
Pulse length	400 s
Number of pulses	~ 30.000

ITER goals:

- **Show scientific and technological feasibility of fusion energy for peaceful purposes.**
- **Test essential technologies in reactor-relevant physics and technology environment.**
- **Demonstrate safety and environmental acceptability of fusion.**

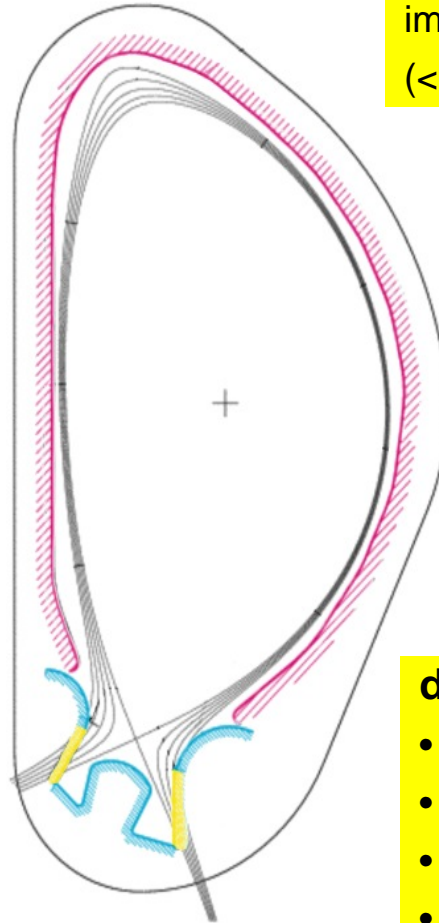
Loading of materials in a fusion device

structural materials

- *thermomechanical loads*
- *electromagnetical loads*
- *neutron irradiation*

plasma facing materials

heat sink materials



bulk plasma:

impurity tolerance
($<10^{-5}$ W, 10^{-2} Be, C)

tritium inventory:

- to be kept low (safety)

divertor target:

- stationary high heat flux 10 (20) MW/m²
- transient heat loads: e.g. disruptions
- highly loaded surface approx. 50 m²
- neutron damage: < 0.5 dpa

Plasma facing materials

first wall: **beryllium:** *low Z*

divertor: **3d C-C composite:**

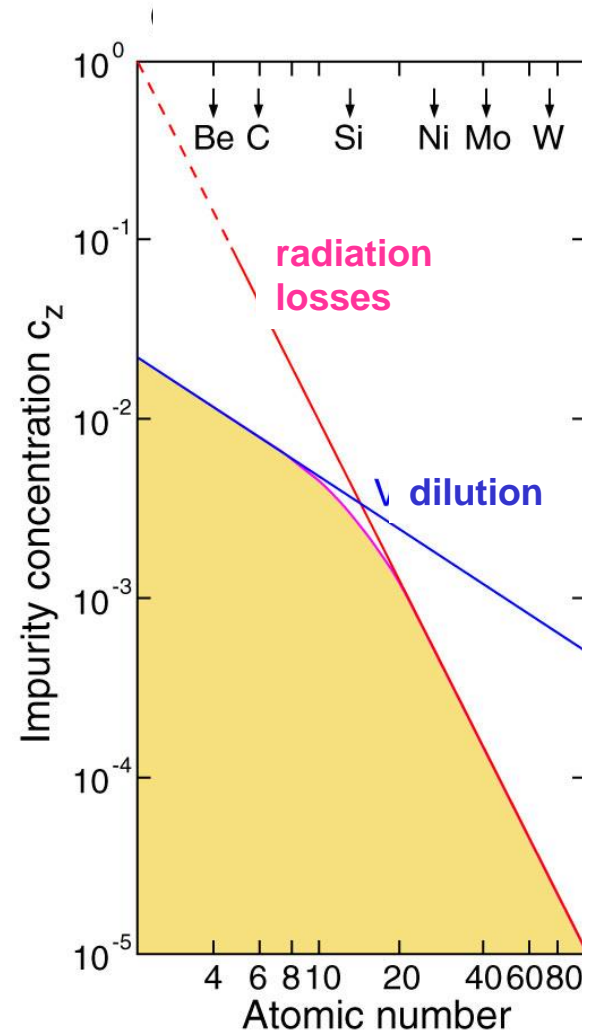
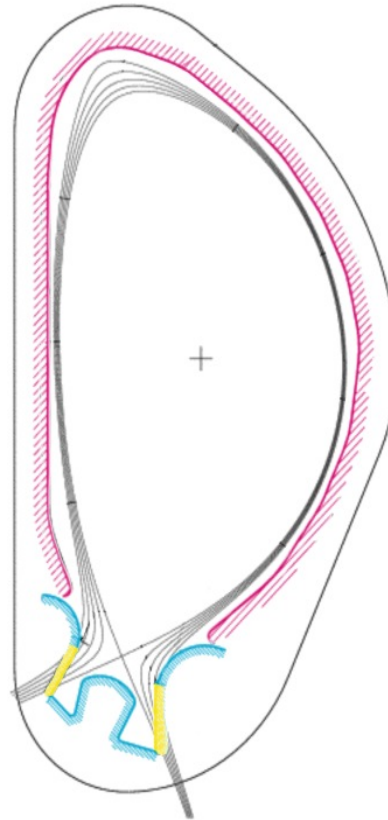
thermal shock resistance

tungsten:

high temp. resistance

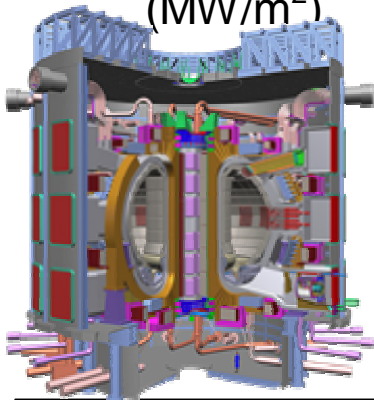
→ materials selection driven by T-codeposition and thermal transients

Heat sink material
CuCrZr

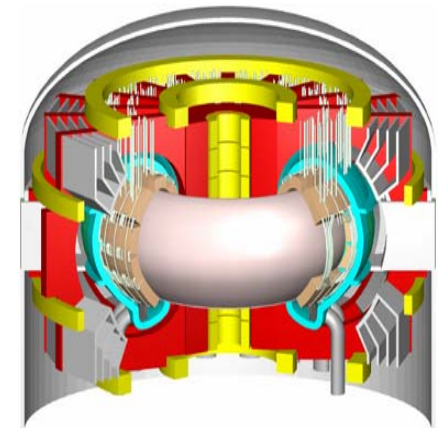


Loading conditions for the divertor

	Divertor target ITER	Divertor target Reactor (DEMO)
component replacements	up to 3	5 year cycle
<u>av. neutron fluence</u> (dpa)	max. 0.5	30
<u>Normal operation</u>		
No. of cycles	10000?	<1000
coolant temperature (°C)	100	300 (600, He)
Surface heat flux (MW/m ²)	10 (20)	10...15



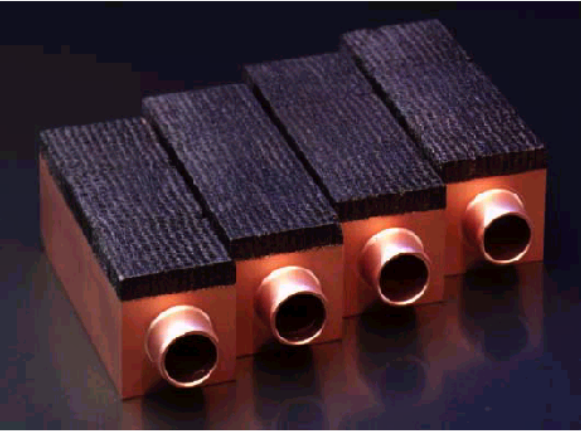
ITER



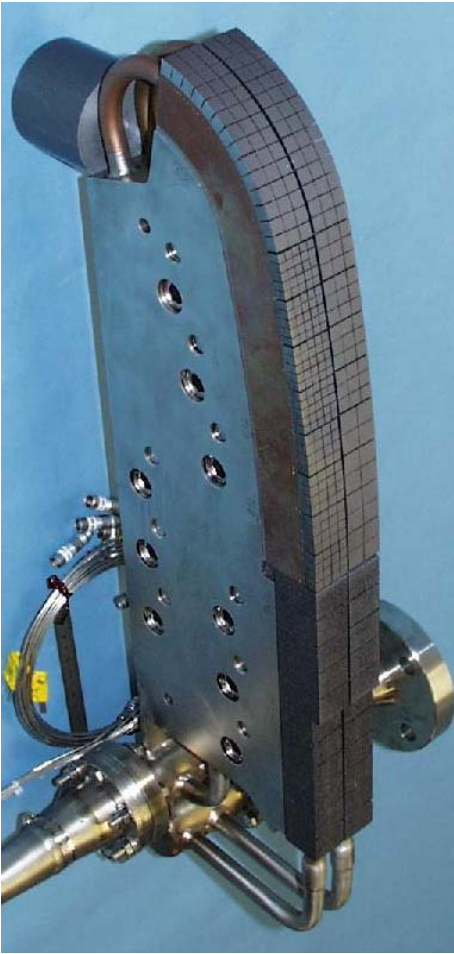
Reactor

ITER Divertor cassette

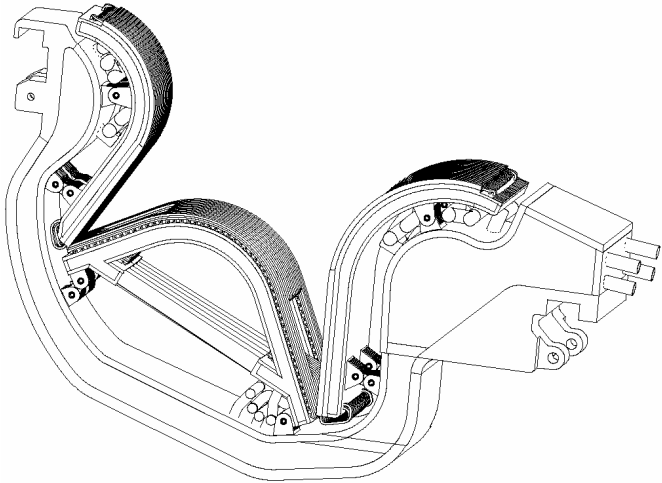
Plasma facing armour:
Tungsten and CFC



25 mm

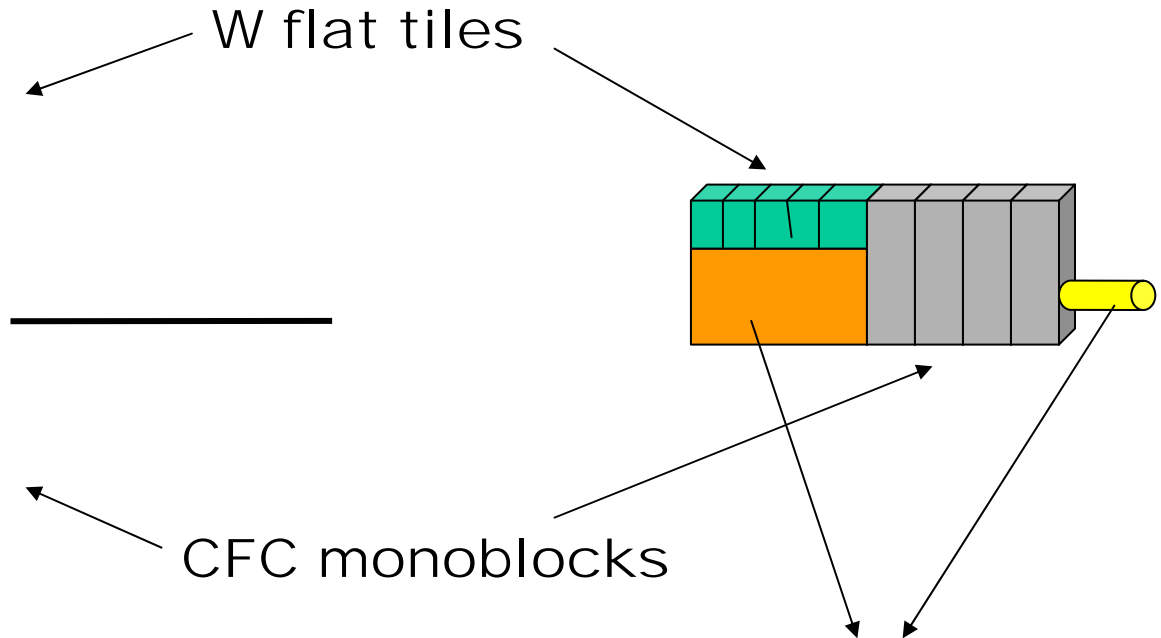
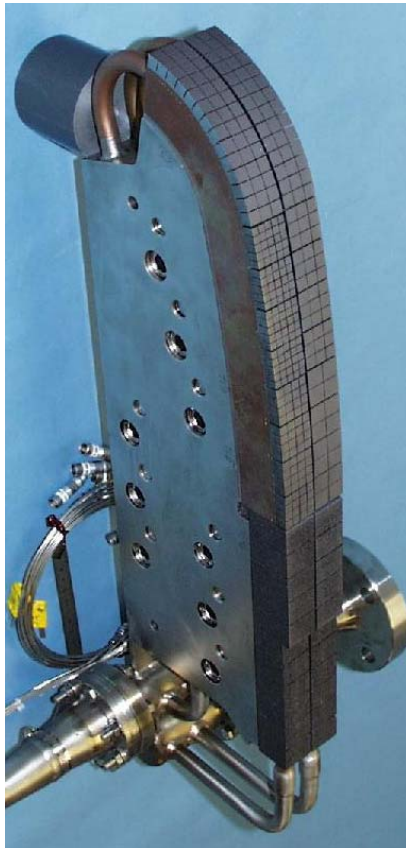


150 mm



400 mm

ITER: Two designs: „Flat tile“ and „Monoblock“



Heat sink material
CuCrZr (ITER)

CuCrZr alloy:

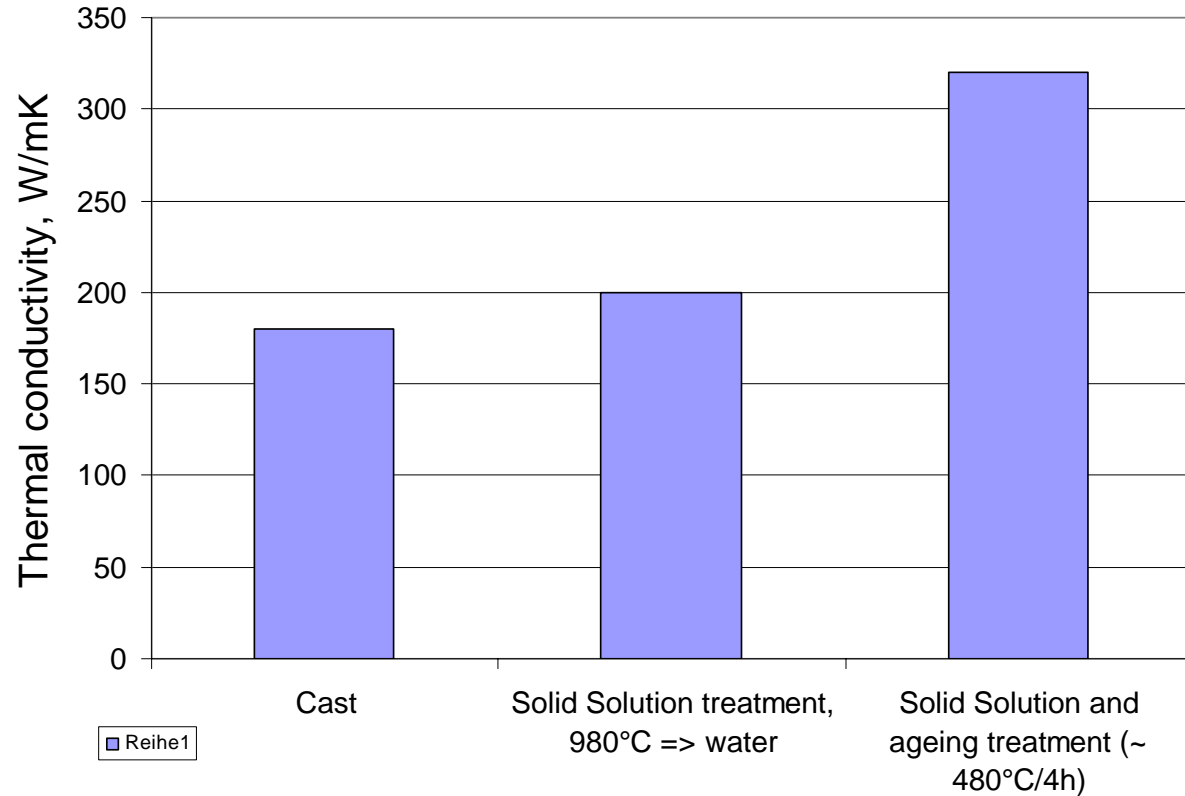
- properties depend largely on heat treatment (manufacturing of components).

<i>CuCrZr</i>	<i>Comments</i>
CHEMICAL COMPOSITION	Detail from Industry
SPECIFIC HEAT	Generally OK
THERMAL CONDUCTIVITY	Generally OK, depends on thermal treatment
THERMAL EXPANSION	Well documented
ELECTRICAL CONDUCTIVITY	Well documented
DENSITY	Well documented
ULTIMATE TENSILE STRENGTH	Generally OK, depends on thermal treatment
YIELD STRENGTH	- " -
ELONGATION	- " -
REDUCTION OF AREA	- " -
ENGINEERING STRESS-STRAIN	- " -
YOUNG'S MODULUS	Well documented
STRESS RUPTURE	Some data exists
CREEP AT 1%	Some data exists
FATIGUE	Some data exists, depends on creep
FRACTURE TOUGHNESS	Some data, better than DS Cu

CuCrZr alloy

properties

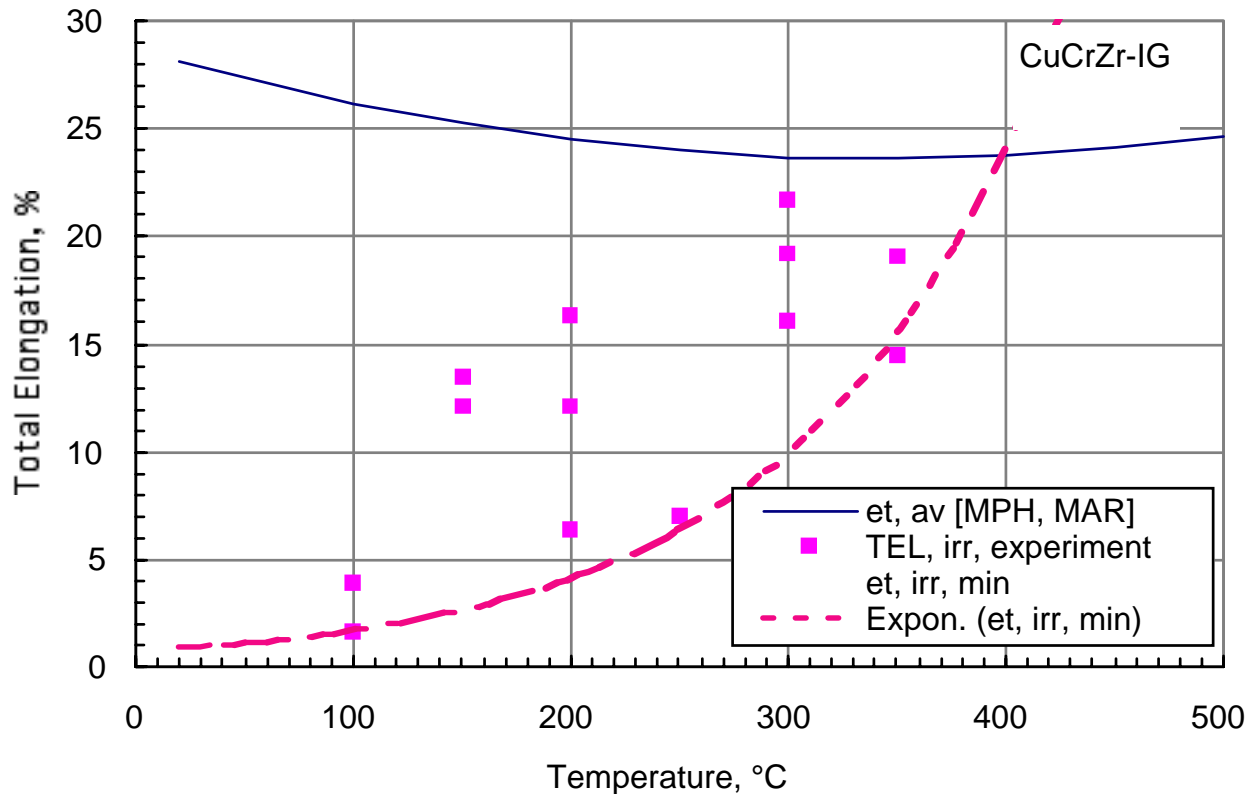
CHEMICAL
SPECIFIC
THERMAL
THERMAL
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YIELD STRENGTH
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YOUNG'S MODULUS



STRESS RUPTURE	Some data exists
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FRACTURE TOUGHNESS	Some data

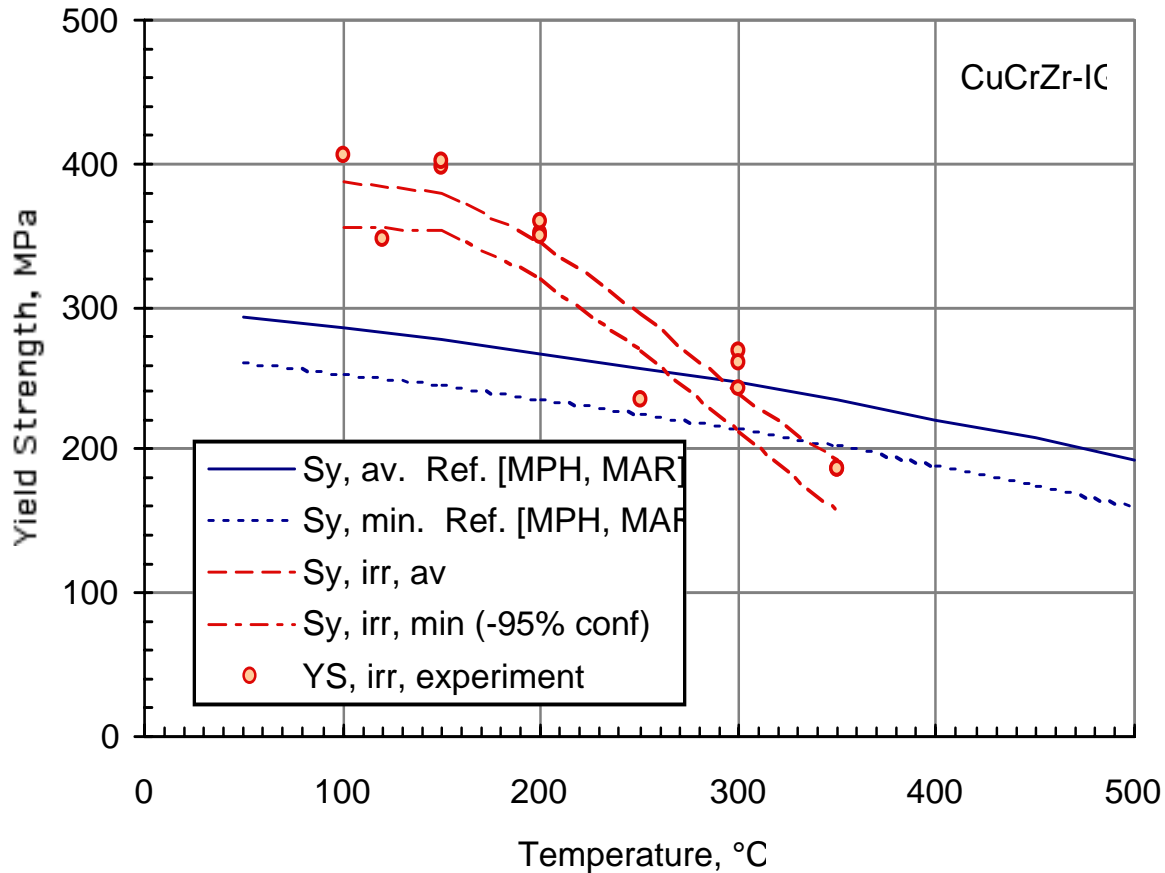
CuCrZr Alloy, Neutron effects

- For ITER conditions: no change of thermal properties,
- However, loss of ductility is main concern for ITER (low operation temperature), but should pose no problem for reactor conditions (high operation temperature)



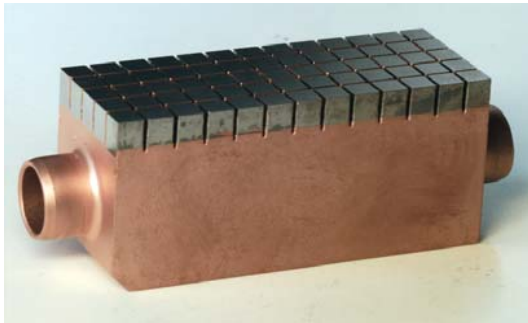
CuCrZr Alloy, Neutron effects:

- For reactor conditions irradiation induced creep is main issue



**Dose - 0.3 - 5 dpa,
Tirr = Ttest**

Tungsten Macrobrush Mock-Ups



Unirradiated

- 1000 cycles x 8 MW/m² – no failure
- 1000 cycles x 14 MW/m² – no failure

200°C, PARIDE 4 (0.5 dpa in tungsten)

- 1000 cycles x 10 MW/m² – overheating
- 1000 cycles x 14 MW/m² – loss of tiles

CFC Monoblock Mock-Ups



Unirradiated

- 1000 cycles x 19 MW/m² – no failure
- 700 cycles x 23 MW/m² – no failure,

200°C, 0.2 dpa (in carbon)

- 1000 cycles x 10 MW/m² – no failure
- 1000 cycles x 12 MW/m² – no failure
- screening at 14 MW/m² – surface erosion

Activity: FZ Juelich and HFR, Petten

Comparison of PH alloys and DS Cu

PH Cu alloys	DS Cu alloys
<i>Thermal stability</i>	
Above ageing temperature overageing: significant decrease of strength. Overageing affects also the thermal conductivity by the dissolution of precipitates.	Inert alumina particles are not prone to coarsening or to dissolution, keeping their hardening effect up to very high temperatures. Properties strongly depend on the production route and are less sensitive to heat treatments.
<i>Fracture toughness</i>	
FT of unirradiated and irradiated materials decreases with increasing temperature, but remains at a relatively high level.	Very low above 200°C in the unirradiated condition. Fracture toughness of irradiated GlidCop Al25 decreases 2-3 times compared to unirradiated material.
<i>Isotropy</i>	
Isotropic mechanical properties.	The short-transverse ductility and fracture toughness is less than in the other two directions.
<i>Weldability</i>	
Can be welded by TIG and EB and then solution annealed and aged without cold work, recovering 50-70% of the full hardened strength.	Not suitable for structural/leak tight fusion welds. Microstructure is completely destroyed in this case, with unrecoverable loss of strength of the joint. Non-fusion weld should be applied (friction, explosion, etc.).
<i>Neutron irradiation resistance at high temperature</i>	
The PH alloy microstructure is less stable under irradiation, due to radiation enhanced coarsening of the Cr/Zr precipitates. Irradiation induced creep at >350°C.	DS alloys have a higher stability range, but are also expected to show irradiation induced creep at high temperature.

Most important: efficient energy conversion

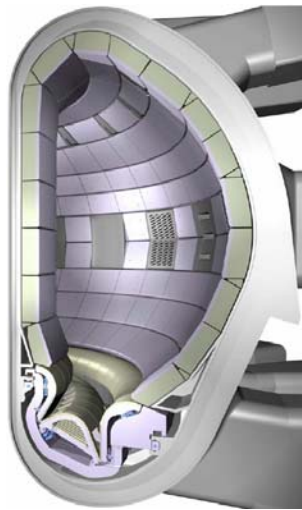
- **Water cooled** divertor: close to PWR conditions, water at 300°C, 10 MPa:
new heat sink materials needed

Attractive, higher thermal efficiency:

- **He-gas cooled** W-based divertor:
advanced technology (min. 600°C He at 10 MPa)
open materials questions
-

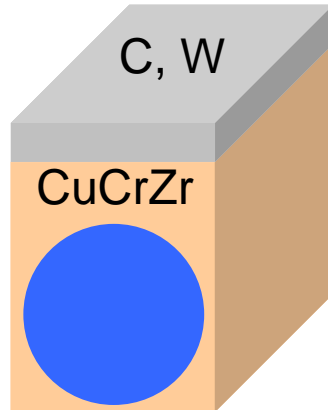
Heat sink materials: ITER – reactor (DEMO)

ITER - Divertor

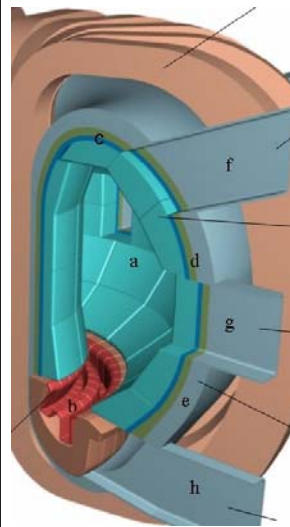


- divertor: 10-15 MW/m²
- coolant: water 80°C
- no energy production
- neutron irradiation ≤ 0.5 dpa
- use of available materials

Heat sink:
CuCrZr
max. operation
temperature:
350-400°C

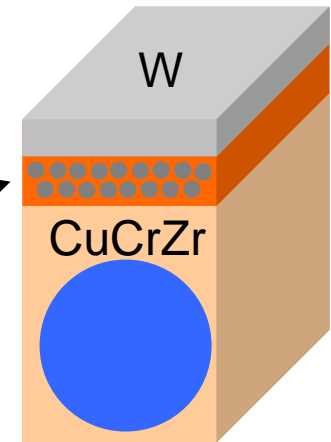


DEMO - Divertor



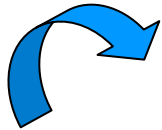
- divertor: 10-15 MW/m²
- coolant: water ≥ 300°C or helium ~ 600°C
- energy production
- neutron irradiation ~ 30 dpa
- development of new materials

Heat sink:
SiC fibre reinforced
copper
operation temperature:
~ 550°C

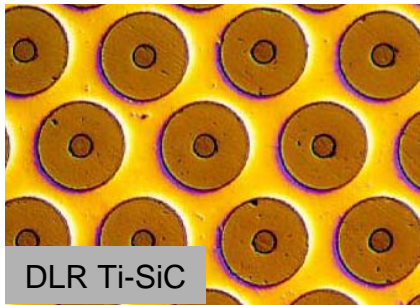


Motivation: Cu-SiC MMCs

aim: composite tensile strength 600-800 MPa at room temperature



important: optimised bonding between the fibre and matrix

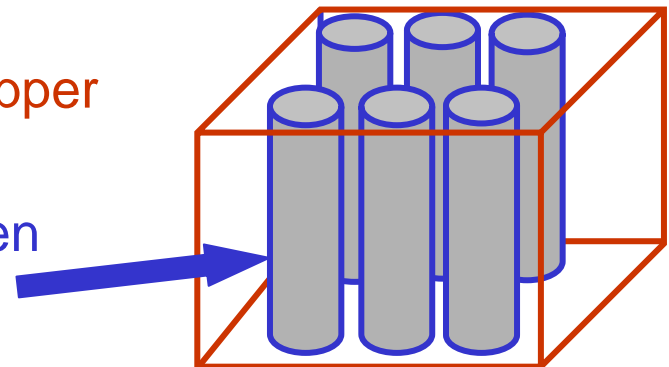


DLR Ti-SiC

- e.g. DLR: titanium matrix composite reinforced with SiC long fibres for aeroplane engines
- interf. shear strength in the range of 70-80 MPa

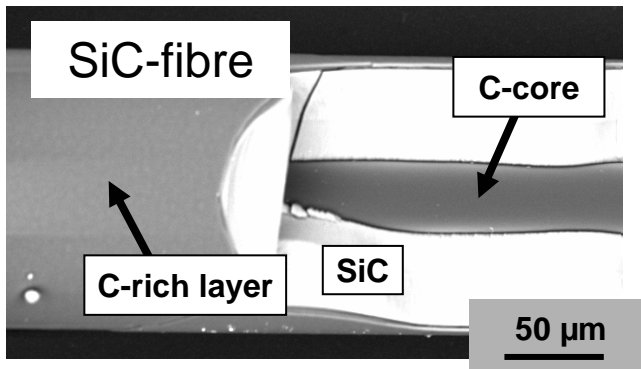
problem: adhesion between SiC/C and copper

solution: titanium interface layer between SiC fibre and copper matrix

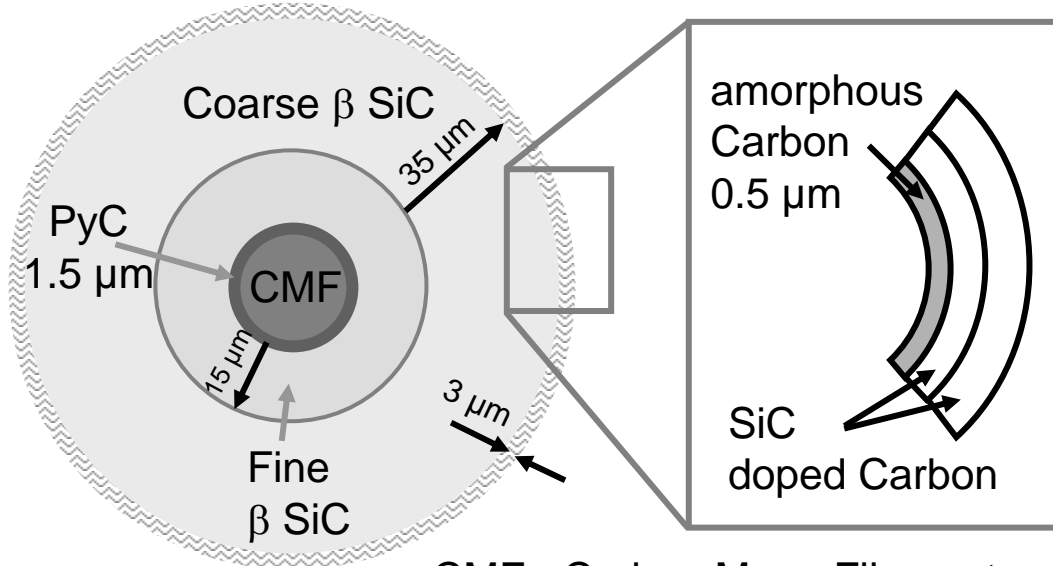
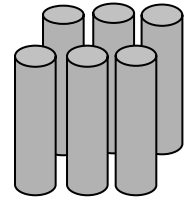


MMC – SiC Fibres

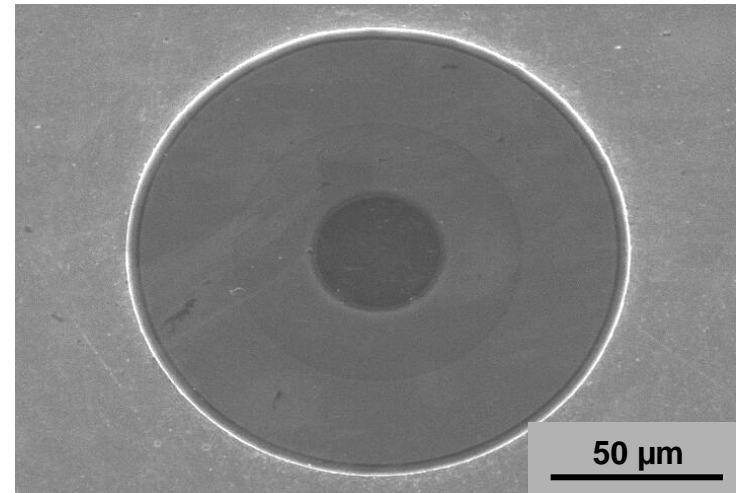
SiC fibre SCS6 (Specialty Materials) \varnothing 140 μ m



- commercially available SiC fibre
- with carbon rich layer at the surface for protection during handling
- optimised for titanium matrix



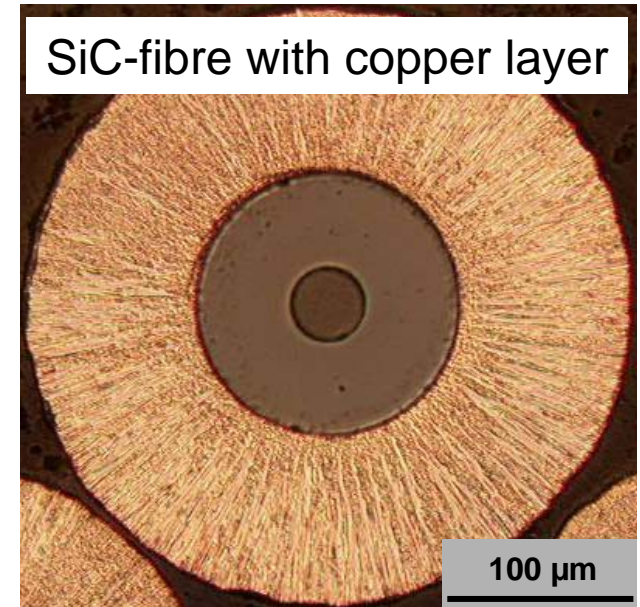
CMF - Carbon Mono Filament



Electroplating of copper



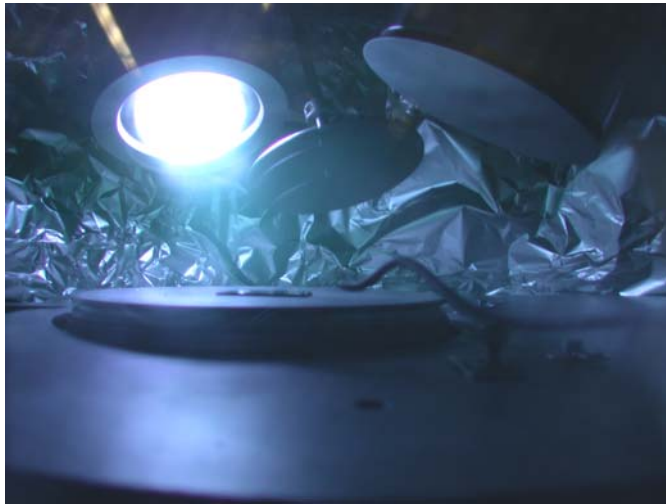
- CuSO₄ bath
- room temperature
- 4,5 V
- 8 hours
- fibre volume fraction $v_f = 20\%$



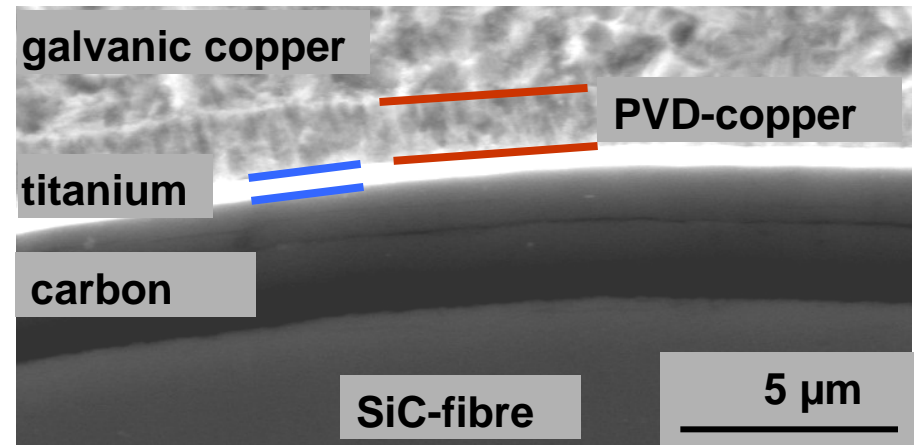
galvanic deposition of a 80 μm thick copper layer as matrix

Processing of MMC – Interlayer

Magnetron sputter device

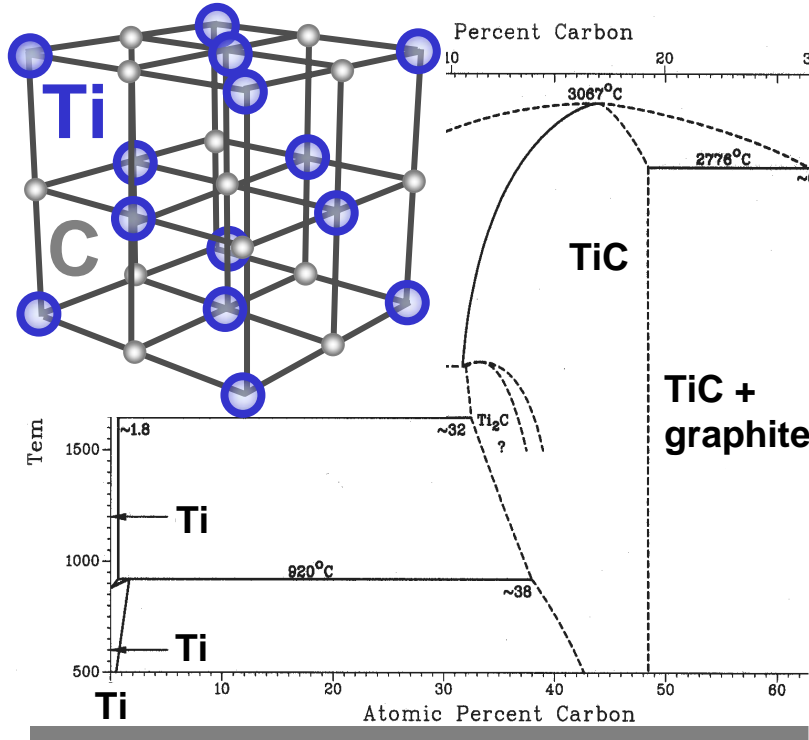


SEM

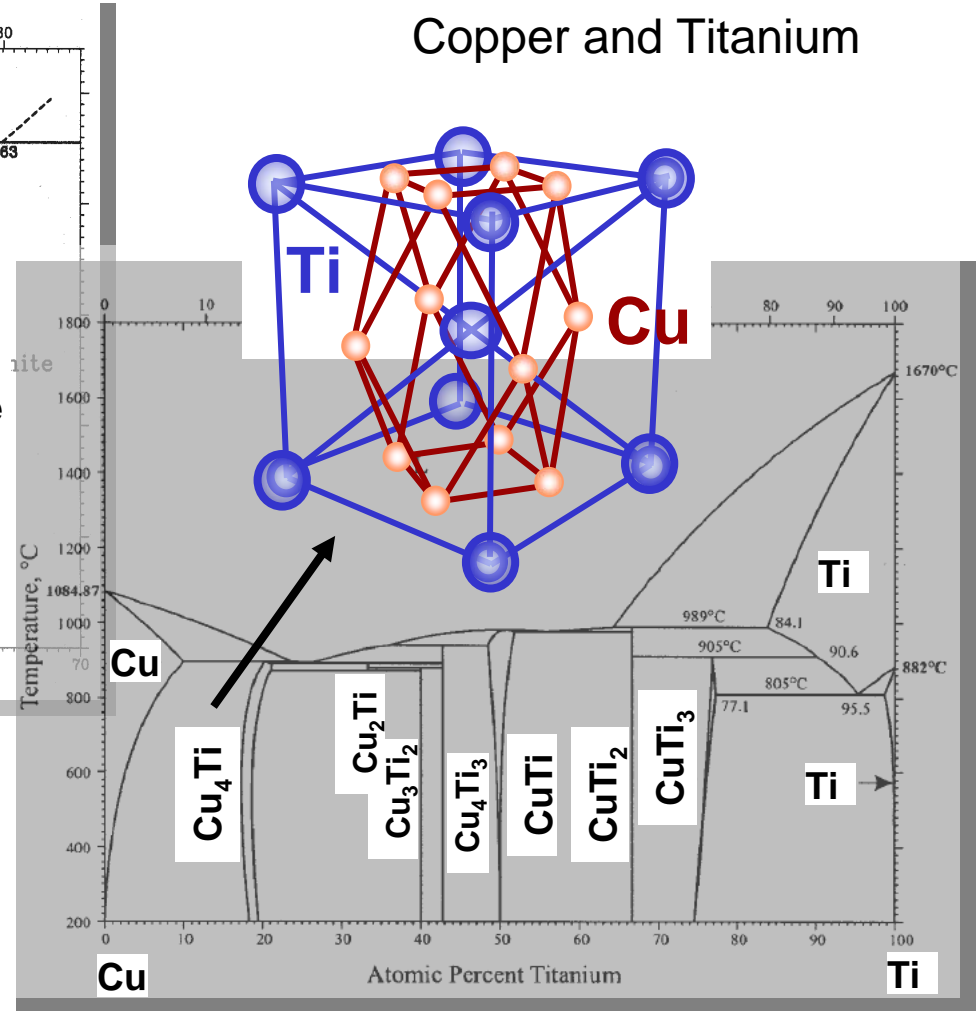


- sputter deposition of titanium interlayer
- layer thickness 100-200 nm
- deposition of copper layer - protective coating (500 nm)

Phase diagrams

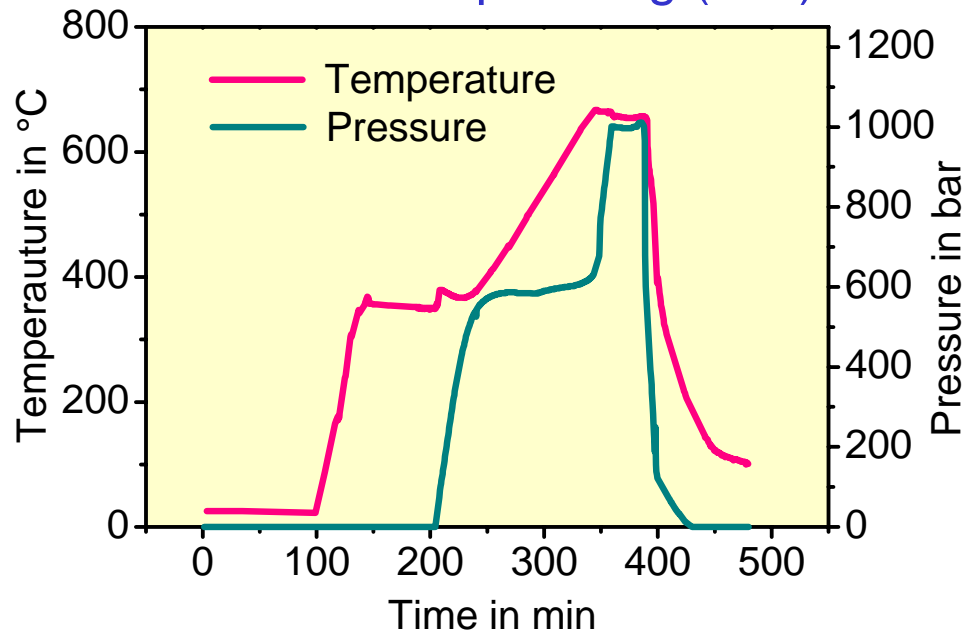


Titanium and Carbon

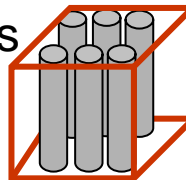


Copper and Titanium

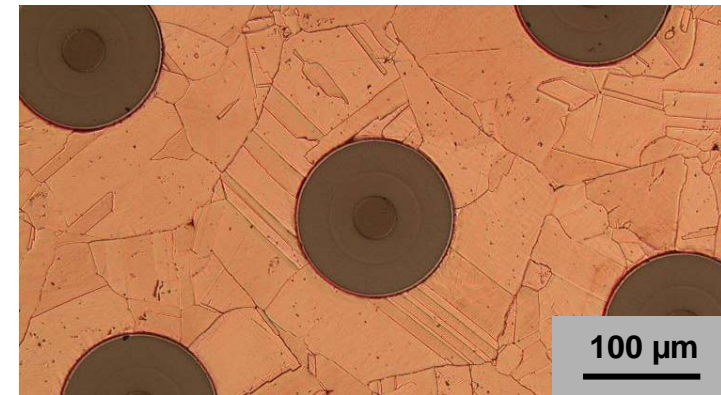
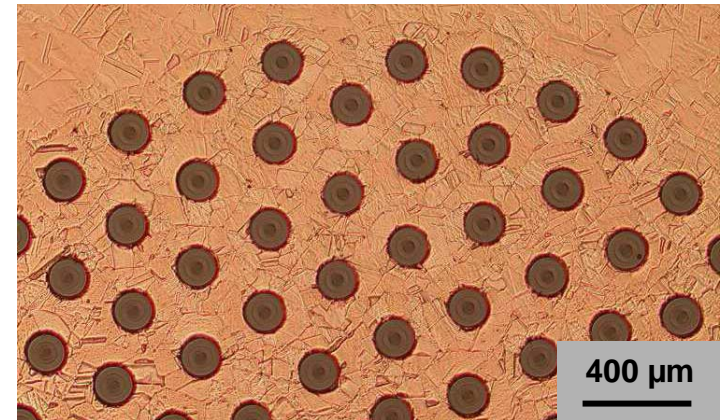
Hot-isostatic pressing (HIP)

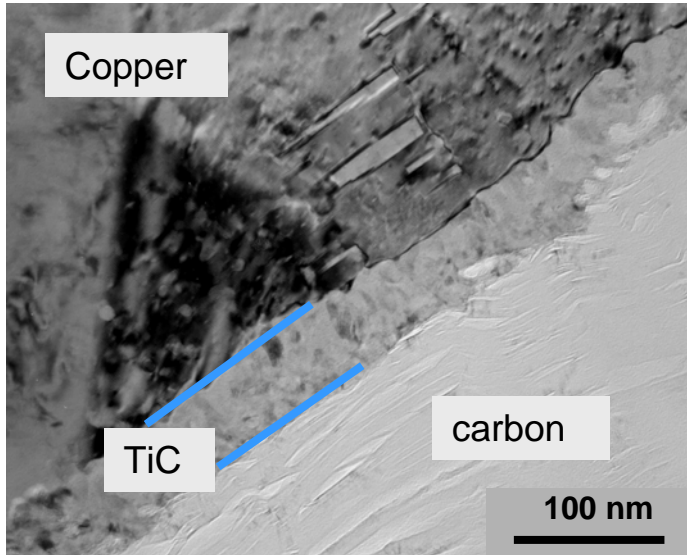


- coated fibres were consolidated in a copper capsule by hot-isostatic pressing at 650°C for 30 minutes
- maximum pressure 100 MPa



Composite



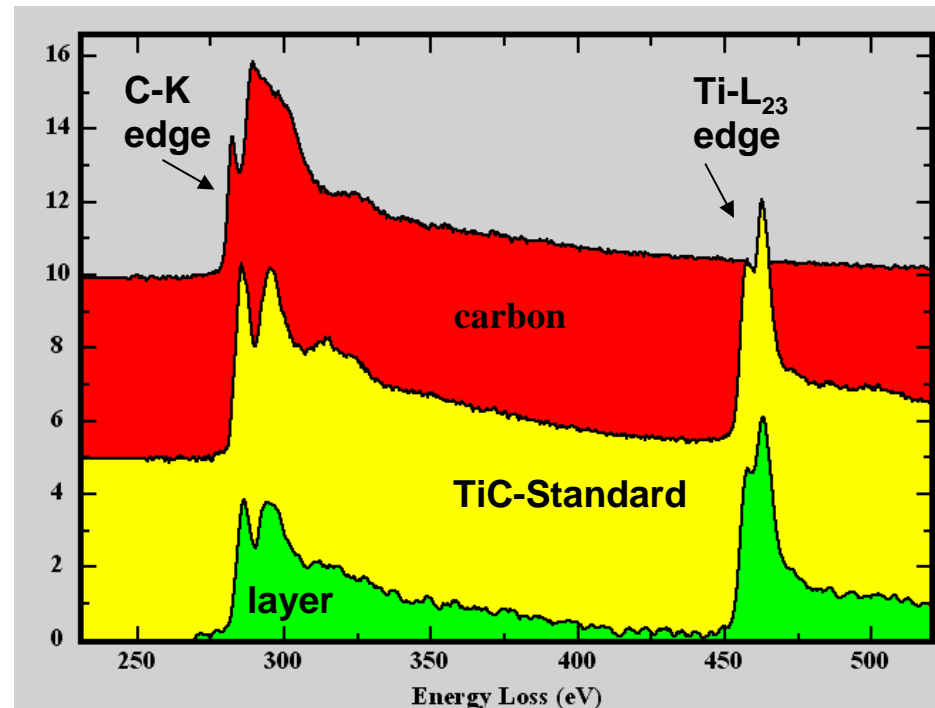


TEM image of interface (PyC)

- EELS: formation of TiC
- formation of a rough interface
- chemical and mechanical bonding between C and Cu

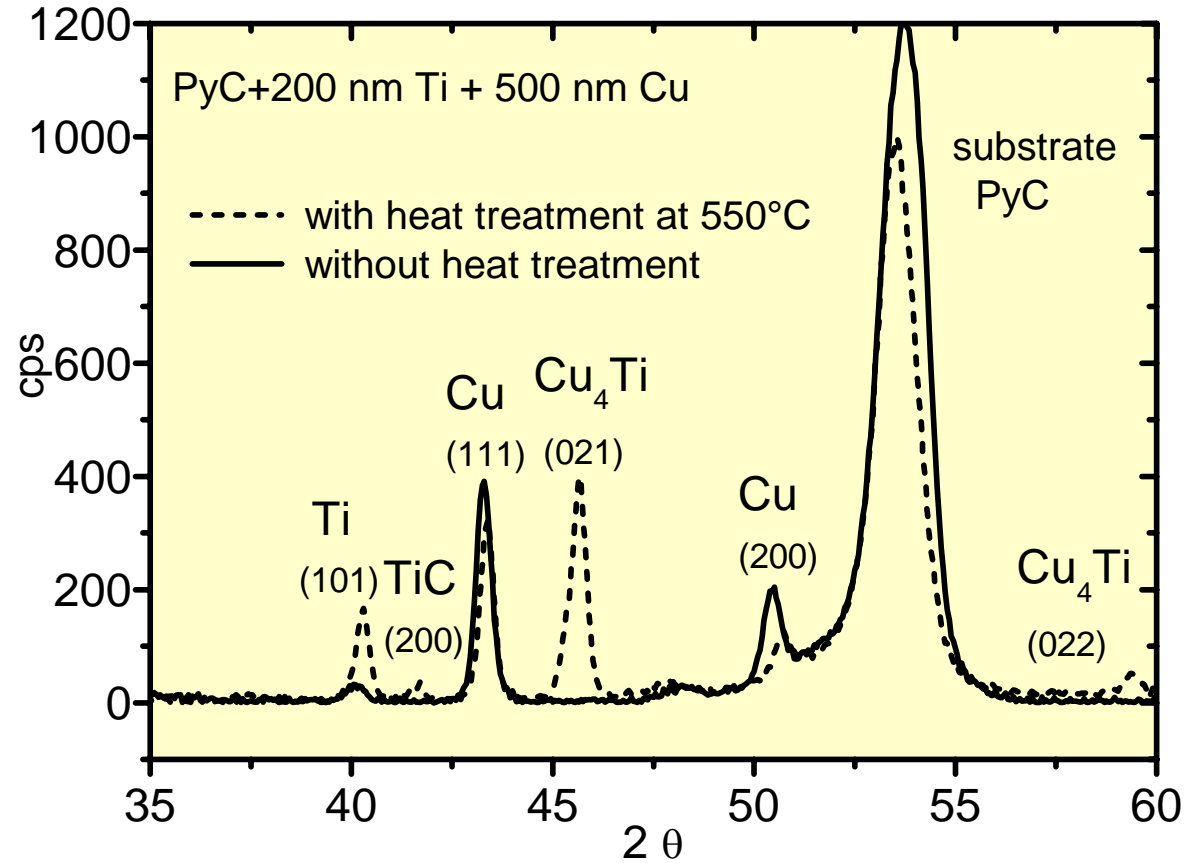
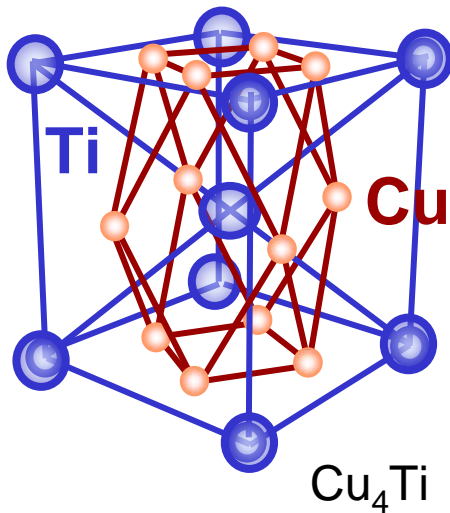
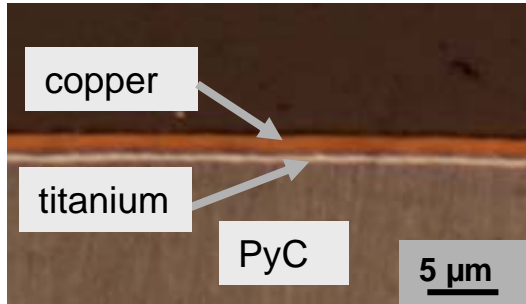
- plane pyrolytic carbon substrate (PyC)
- 100 nm Ti + 500 nm Cu
- heat treatment at 650°C/ 1h

EELS (electron energy-loss spectroscopy)

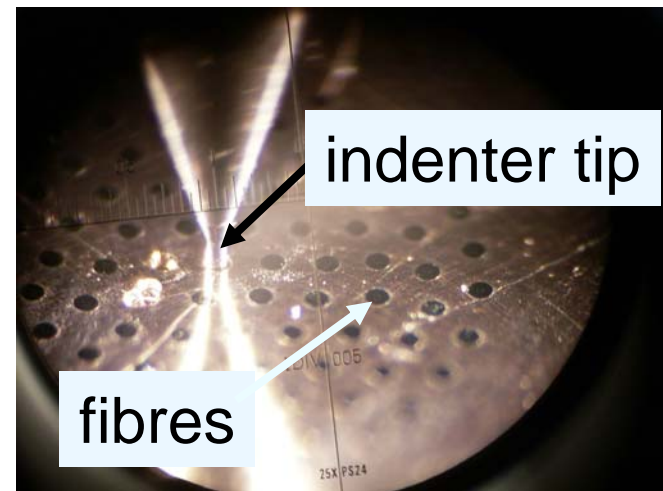
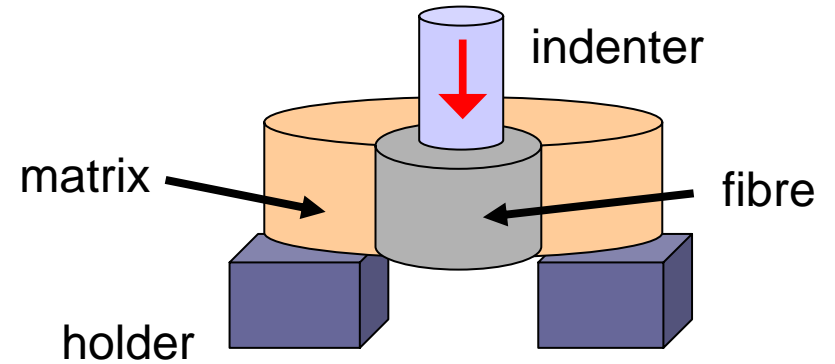
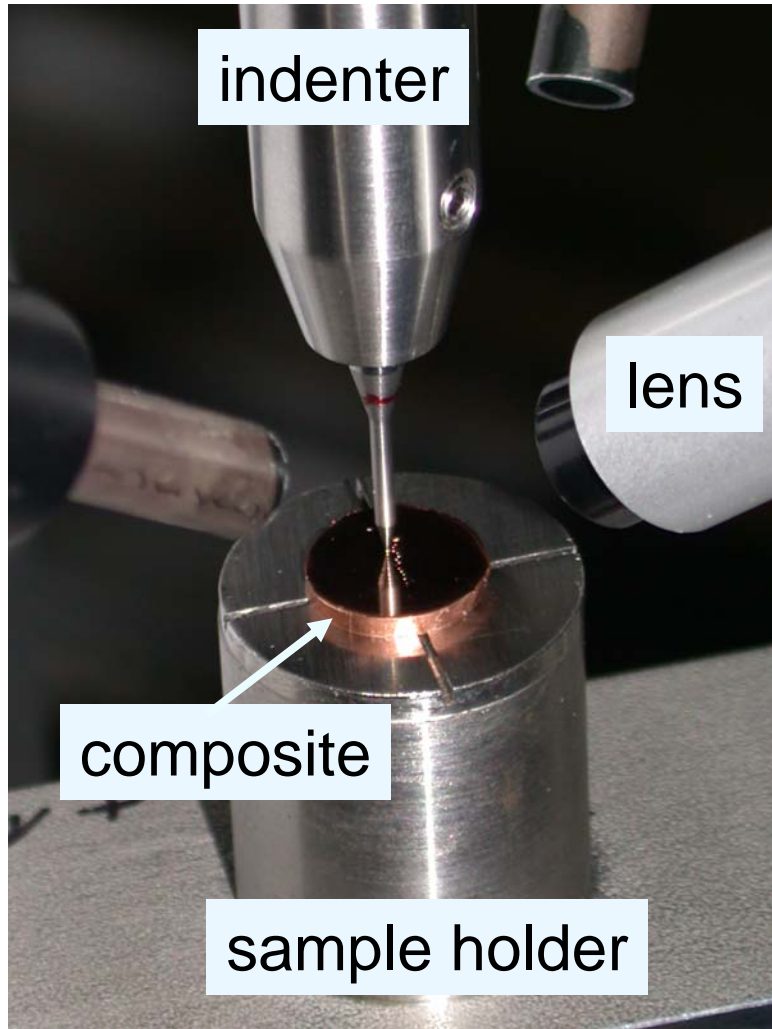


See also Poster by MPI Halle (Woltersdorf, Pippel, Brendel, Bolt), C11

XRD Investigations



Push-Out Test

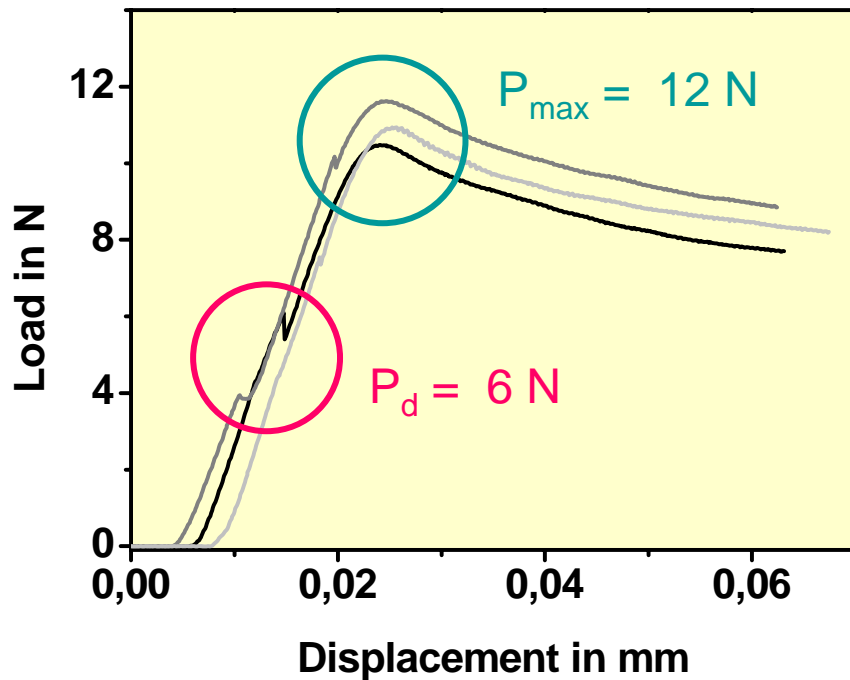


view through magnifying glass

Push-Out Test

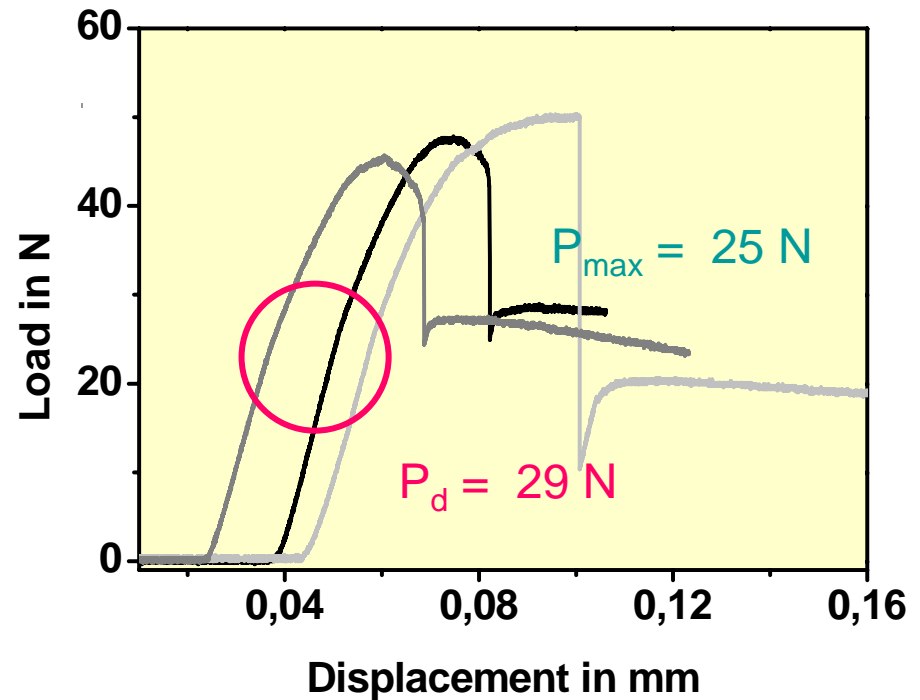
SiC fibre reinforced copper **without** titanium interlayer

Sample thickness 2.4 mm



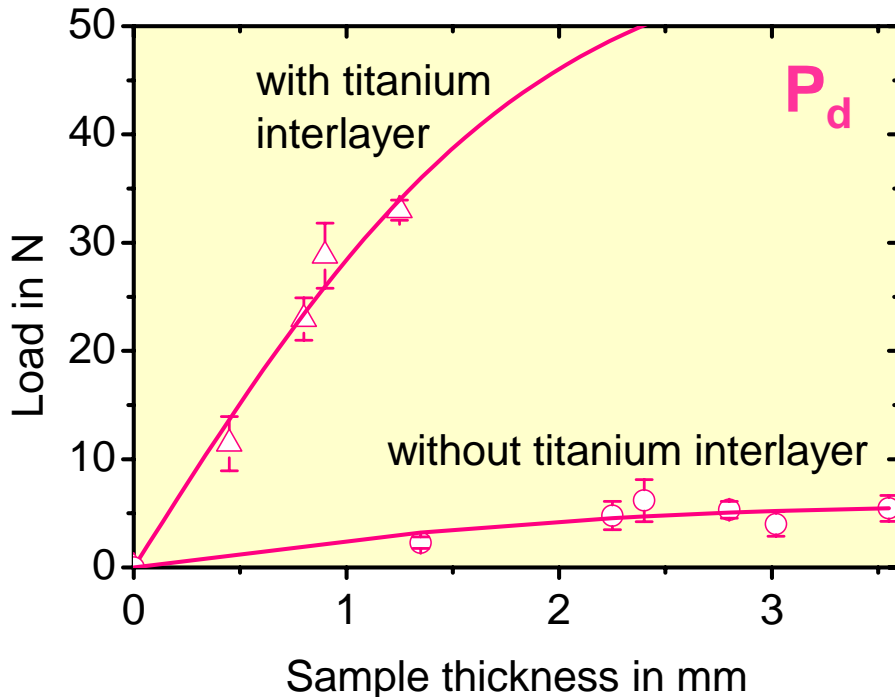
SiC fibre reinforced copper **with** titanium interlayer

Sample thickness 0.9 mm

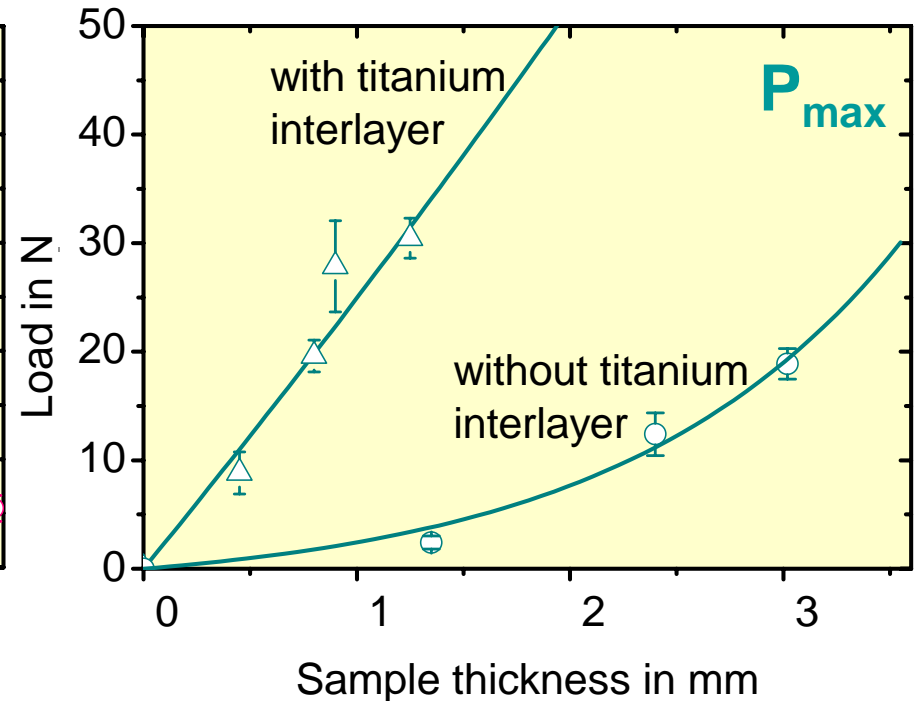


Push-Out Test

Interfacial shear strength



Interfacial friction stress

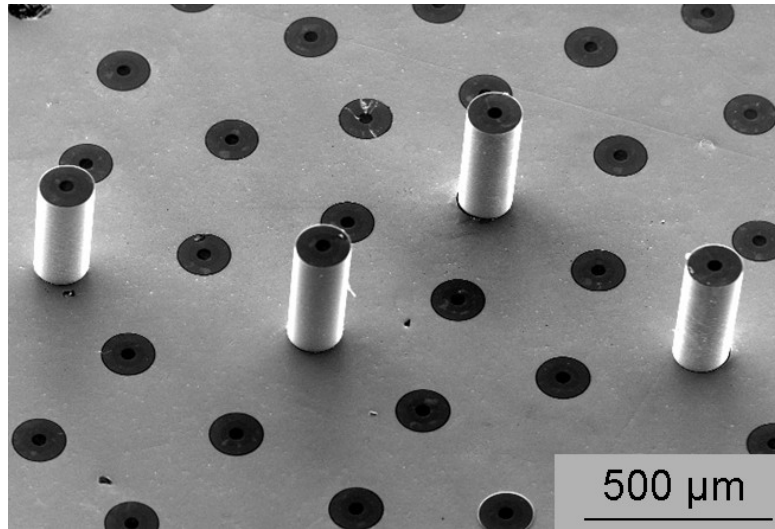


$\tau_d = 6 \text{ MPa}$ without titanium interlayer $\tau_f = 4 \text{ MPa}$

$\tau_d = 70 \text{ MPa}$ with titanium interlayer $\tau_f = 54 \text{ MPa}$

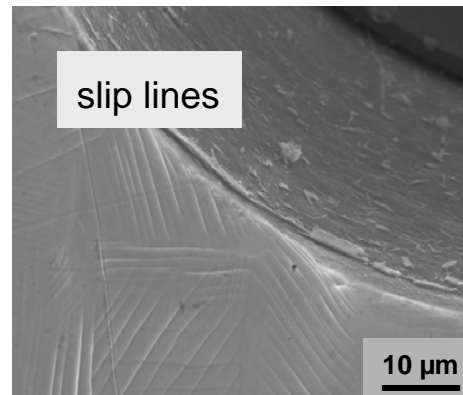
Push-Out Test - SEM

without titanium interlayer

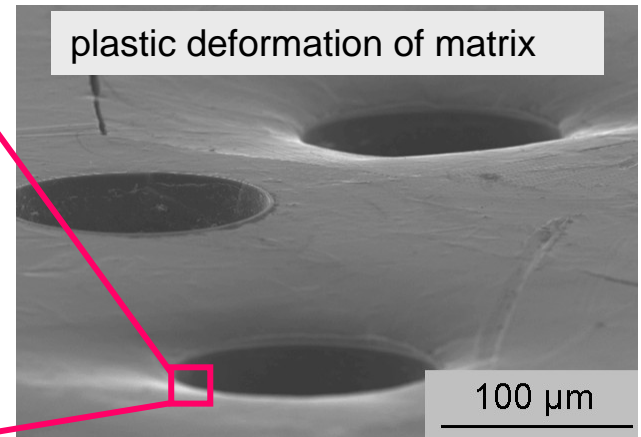
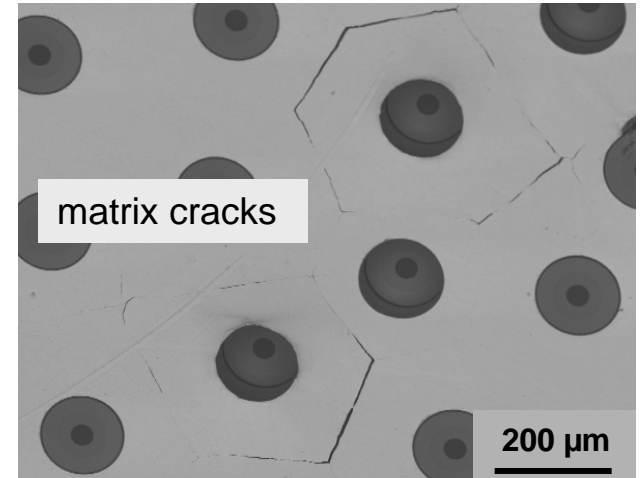


no plastic deformation of matrix after push-out test

front side after push-out test

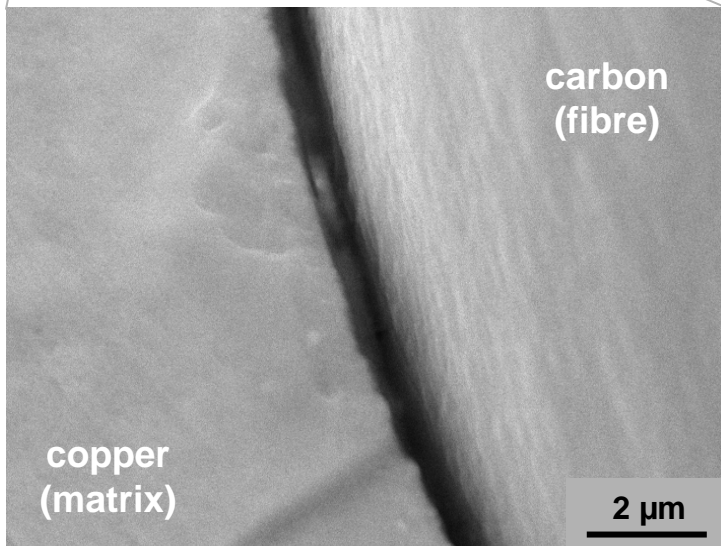
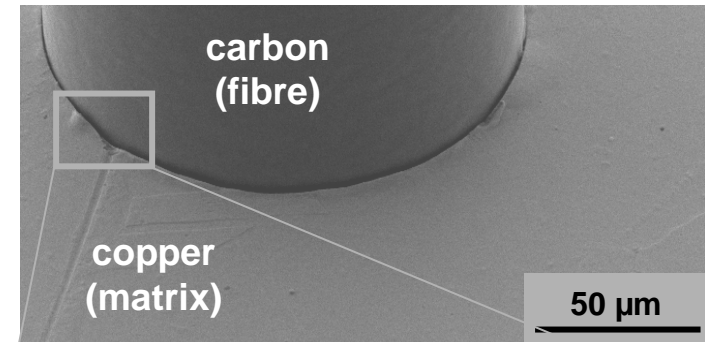


with titanium interlayer

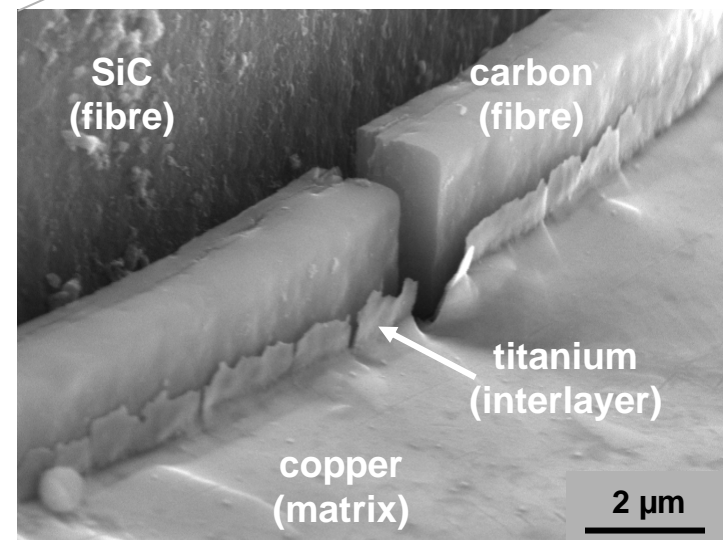
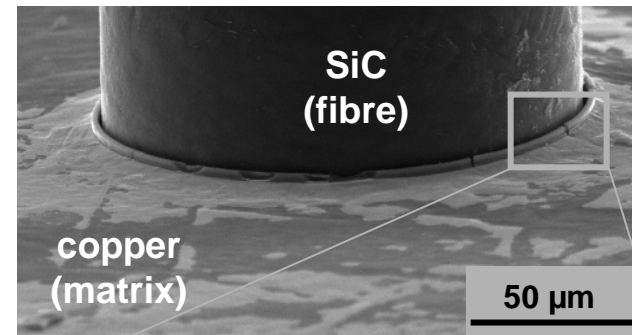


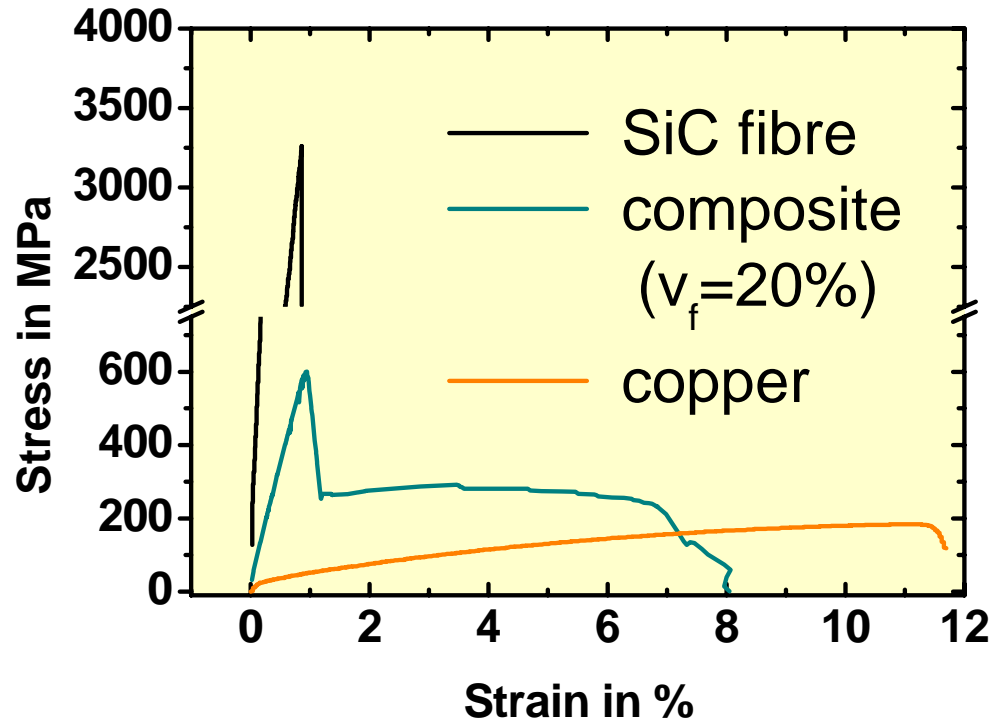
Push-Out Test - SEM

without titanium interlayer



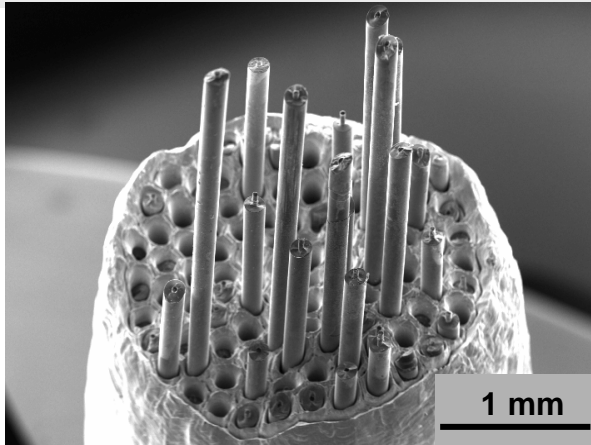
with titanium interlayer





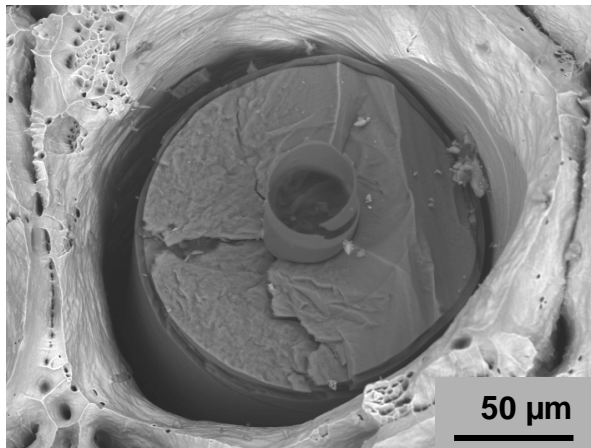
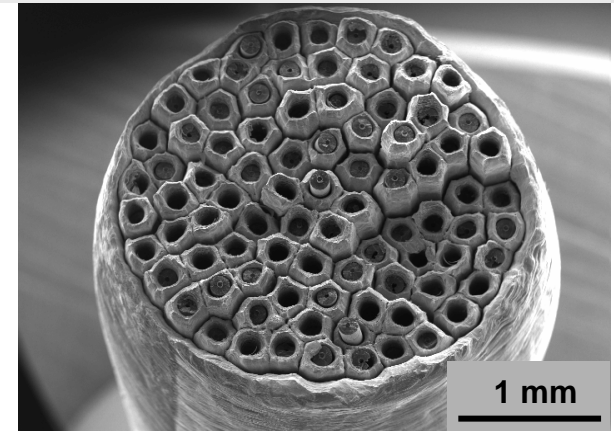
- sample length 70 mm with thread at the ends
- gauge length 10 mm
- diameter in gauge length 3.5 mm (fibre reinforced zone)

Composite without titanium

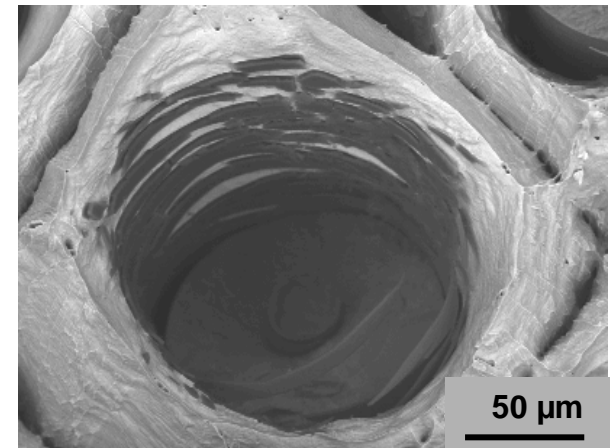


Fibre pull out in the composite without titanium interlayer indicates a weak bonding between fibres and matrix.

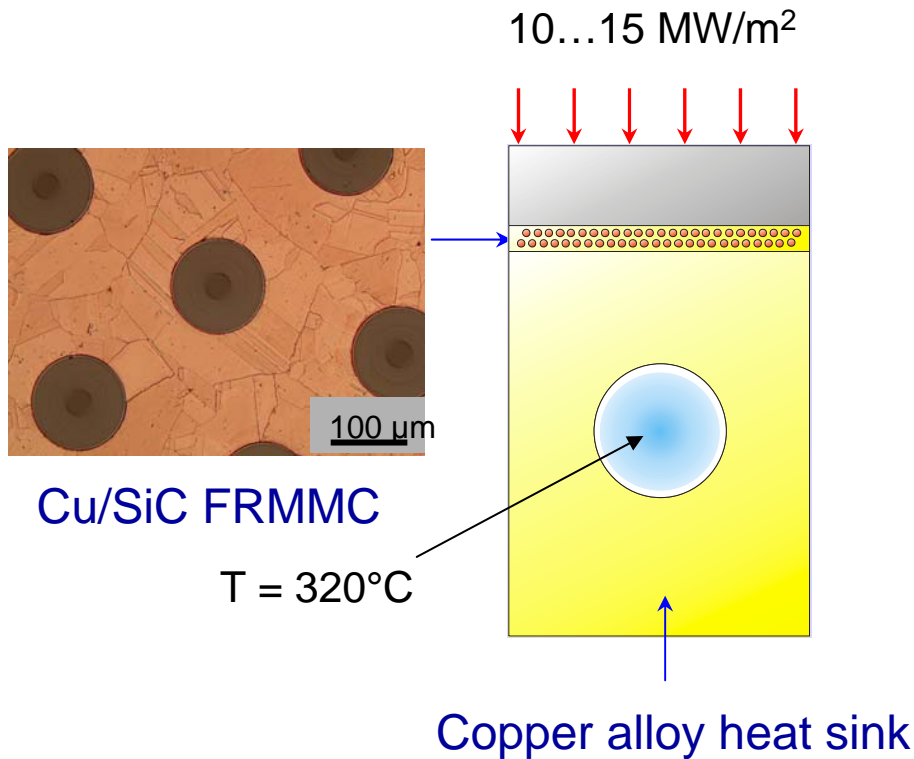
Composite with titanium



Analysis of back scattered electrons shows carbon at the matrix for composites with titanium after tensile test.



Thermomechanics of Cu-SiC FRMMCs



- Open question: plastic stability of the FRMMC under cyclic heat flux loads ?
- Investigation issue: determination of loading limit for plastic instability (shakedown / ratchetting)
- Shakedown analysis; SD limit as design criterion?
- Implications on design

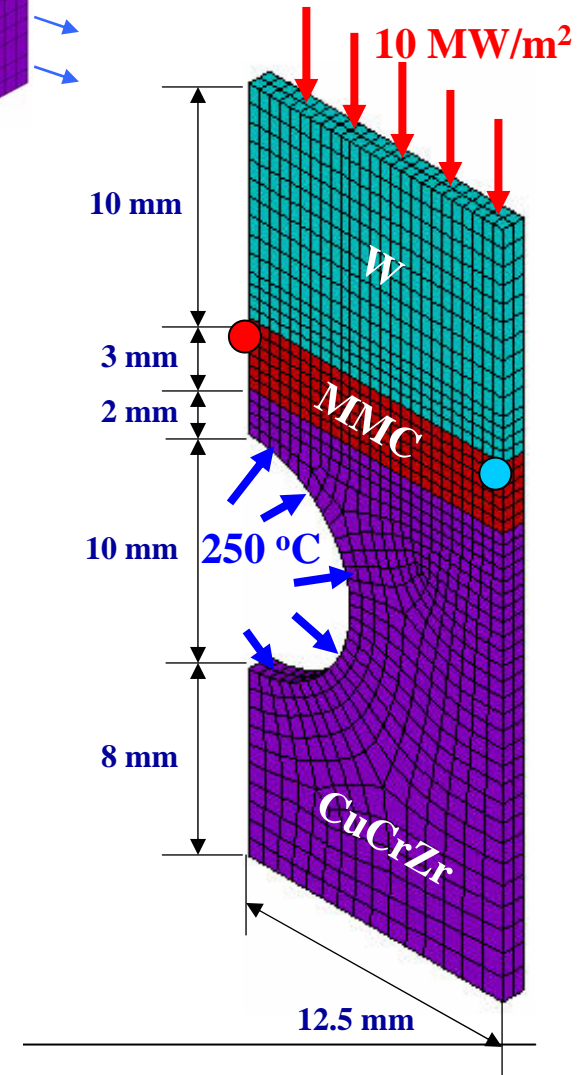
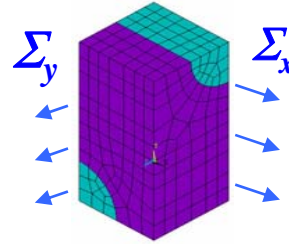
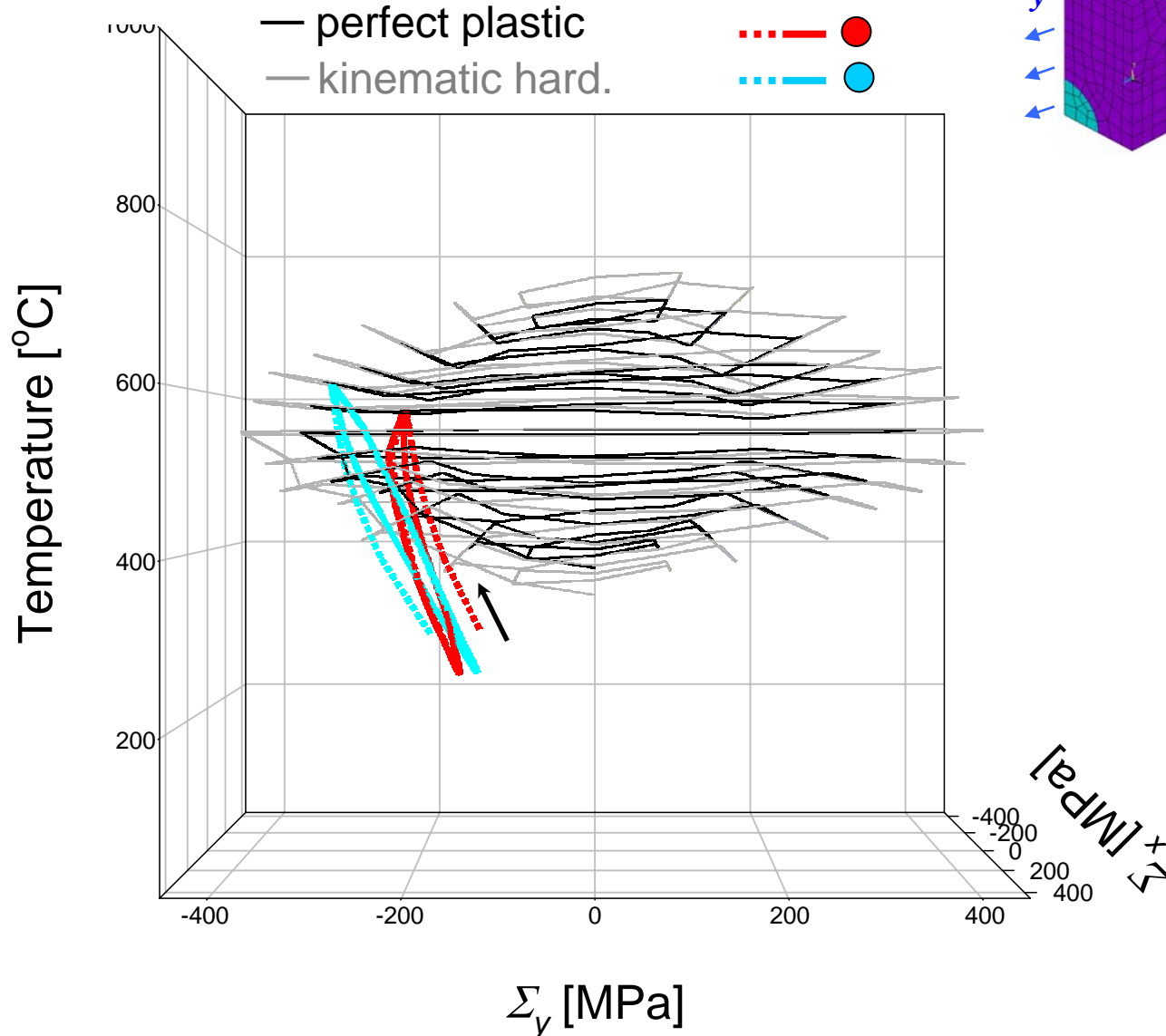
Tensile strength (MPa)
FMMC laminate: 600 (//)
CuCrZr: 400 at RT

CTE ($\times 10^{-6}$)
W: 3.9
FMMC laminate: 12.4 (//)
CuCrZr: 15.5

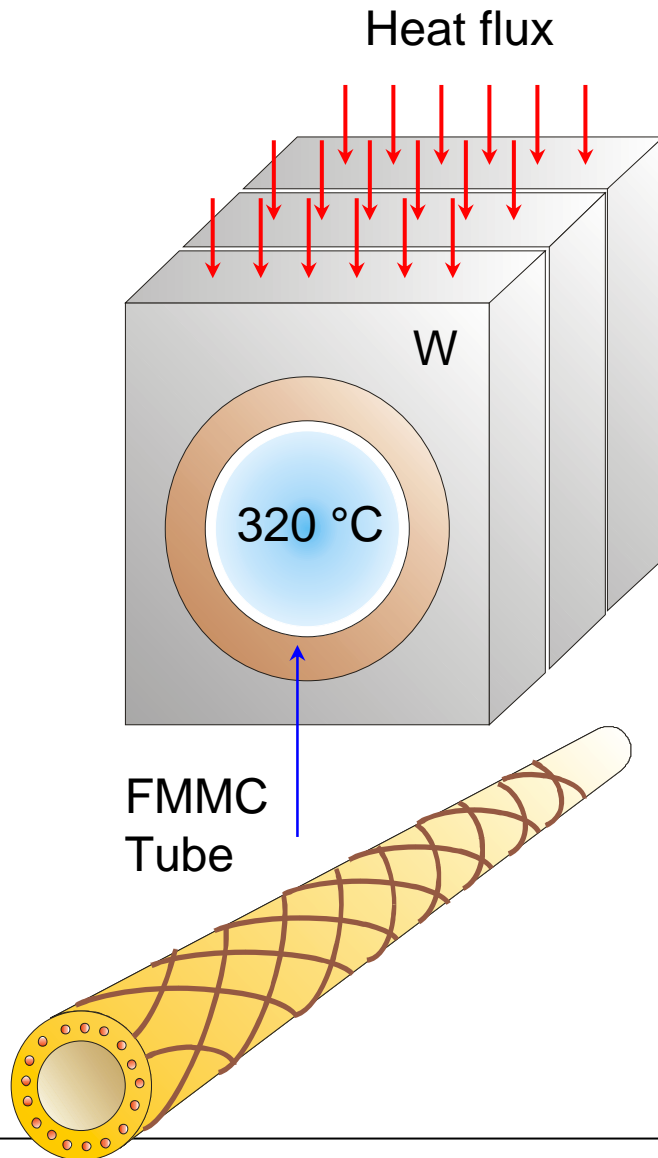
Young's modulus (GPa)
W: 398
FMMC laminate: 165 (//)
CuCrZr: 128

Component thermomechanics under high heat flux loading

Heat flux: 10 MW/m^2 , T_{sf} : 550 C , T_{c} : 250 C



Alternative design: Composite coolant tubes



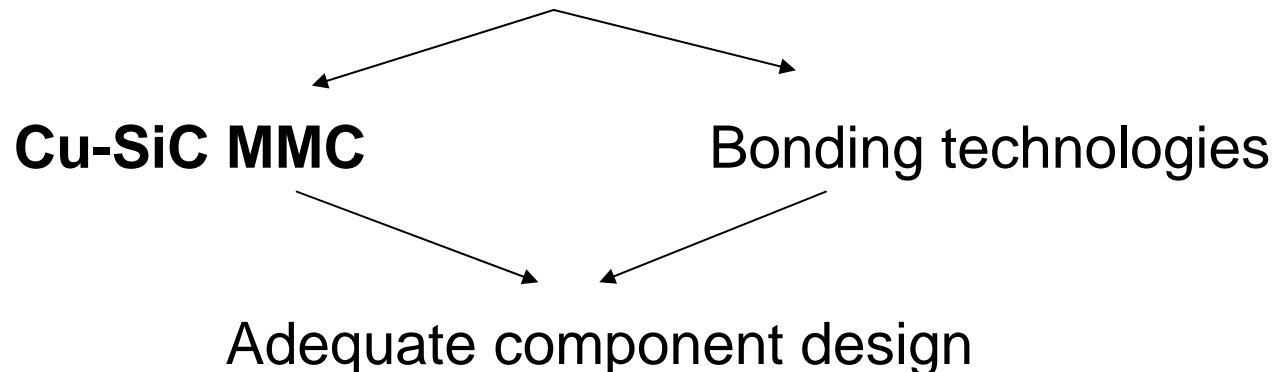
- Merits of this design
 - works with CuCrZr tube for cold ITER loading conditions
 - basic component fabrication technology already developed
- Motivation
 - strengthen the tube for coolant temperature up to 320 °C
 - reduce the thermal stress
- Issues:
 - fabrication of fiber-reinforced composite tubes
 - simulation techniques for 'design by analysis'

Divertor (high thermal conductivity needed):

- CuCrZr: (irrad. data up to 10 dpa)
temperature window:
200°C...350°C
(<200°C: hardening; >350°C: softening)
- DS Cu similar, max. 400°C



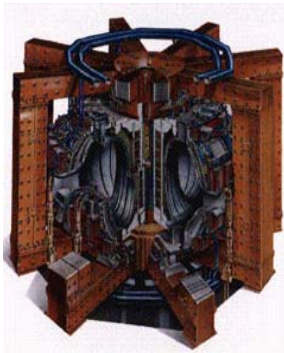
New class of heat sink materials needed
and the respective component technology



Summary: New materials are needed for fusion

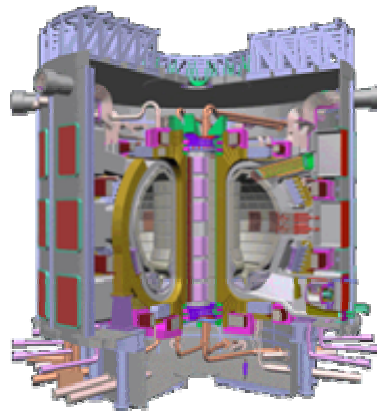
JET

(fusion power 16 MW, 2 s)



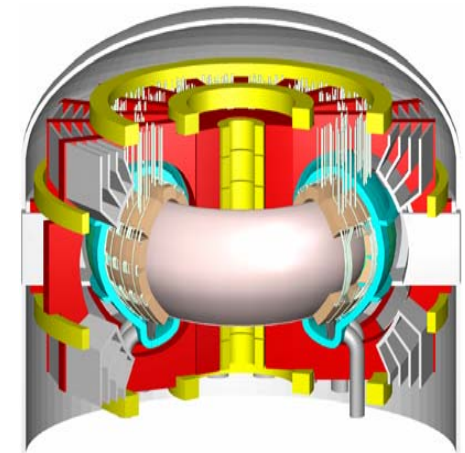
ITER

(fusion power 500 MW, 400 s)



reactor (DEMO)

(fusion power >2000 MW, stationary)



ITER

power reactor

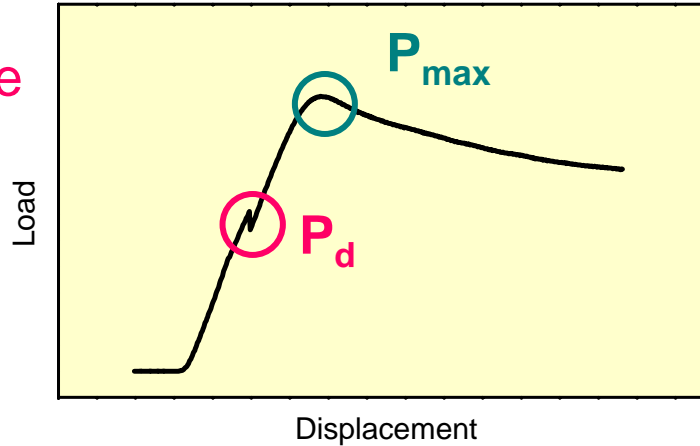
relative size	1	1...1.2
fusion power (MW)	500	2000
power to He-ions (MW)	100	400
total thermal power (MW)		2600
electric power (MW)		1000
efficiency (%)		38
neutron damage (dpa)	5	120 in 5y

Push-Out-Test

P_d – debonding force



Interfacial shear strength τ_d



P_{max} - maximum force



Interfacial friction stress τ_f

$$P_d = \frac{\tau_d \cdot 2\pi R}{\alpha} \cdot \tanh(\alpha \cdot L)$$

○ Fit parameter

$$\alpha = \sqrt{\frac{2G_i}{b_i R E_f}}$$

$$P_{max} = \frac{\pi R^2 \sigma_0}{k} \left[\exp\left(\frac{2\mu k}{R} L\right) - 1 \right]$$

$$\tau_f = \mu \cdot \sigma_0 \quad k = \frac{\nu_f E_m}{E_f (1 + \nu_m)}$$

P_d, P_{max} debonding, maximum force
 R fibre radius
 L specimen thickness

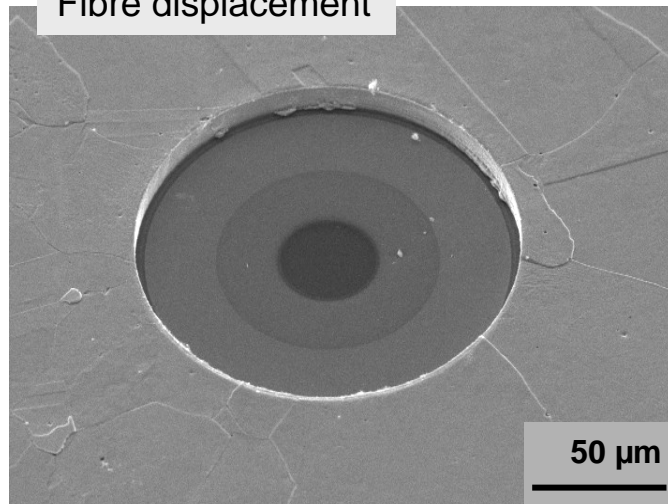
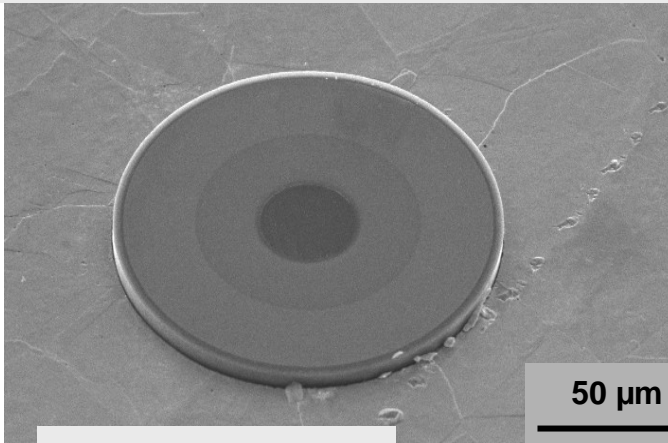
G_i shear modulus
 b_i interface thickness
 E_f, E_m Young's moduli of fibre and matrix

σ_0 radial residual stress
 μ friction coefficient
 ν_f, ν_m Poisson's ratio of fibre and matrix

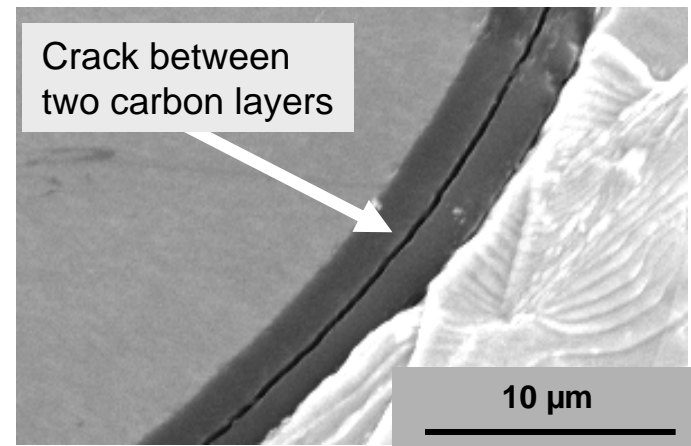
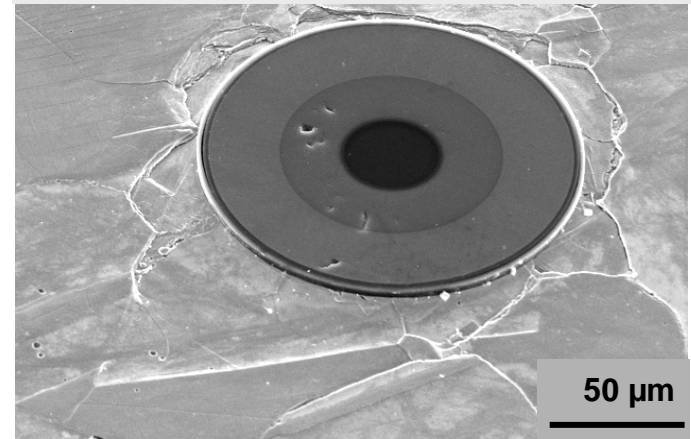
Ref.: Rausch, Meier & Grathwohl, *Journal of the European Ceramic Society* **10** (1992) 229-235

Thermal cycling

Composite without titanium

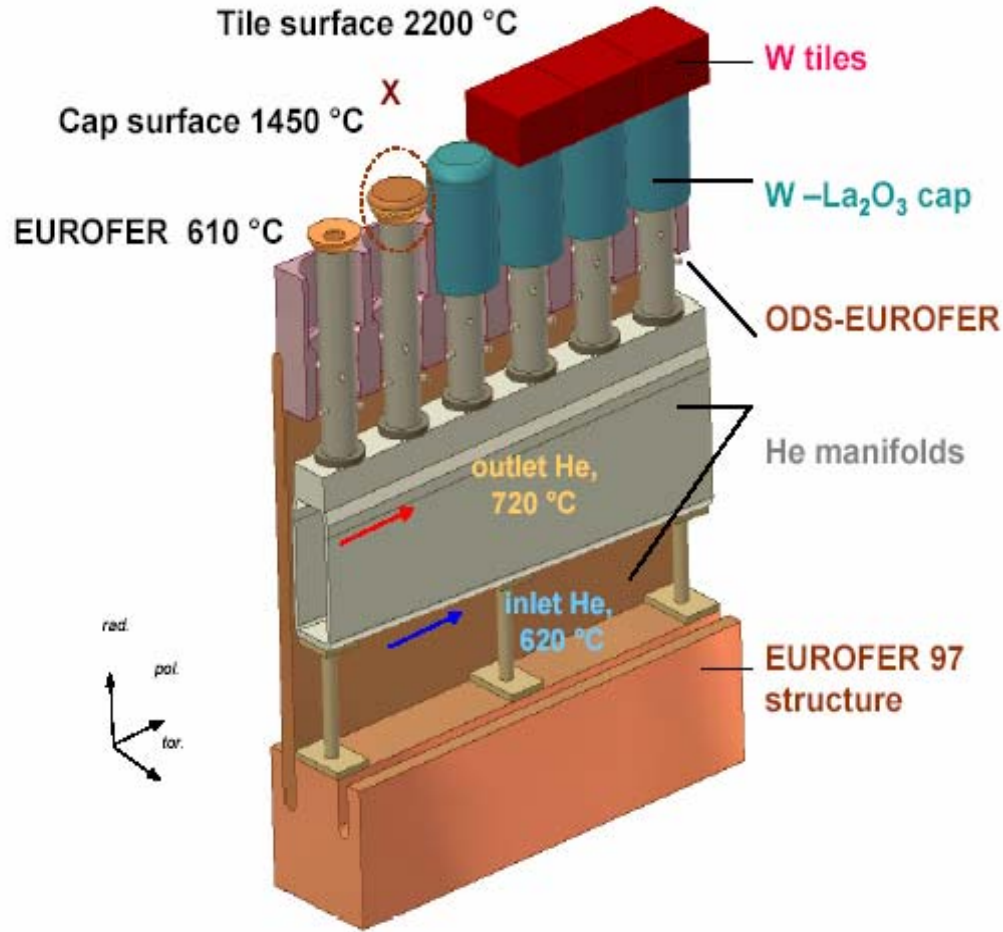


Composite with titanium



120 cycles between 350°C and 550°C

Design of Cooling Fingers in He-cooled Divertor Development



Detail X: Pin plate with outlet tube.

